

## Long term evolution and chaotic diffusion of the insolation quantities of Mars

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Received 21 November 2003; revised 26 March 2004

Available online 2 June 2004

### Abstract

As the obliquity of Mars is strongly chaotic, it is not possible to give a solution for its evolution over more than a few million years. Using the most recent data for the rotational state of Mars, and a new numerical integration of the Solar System, we provide here a precise solution for the evolution of Mars' spin over 10 to 20 Myr. Over 250 Myr, we present a statistical study of its possible evolution, when considering the uncertainties in the present rotational state. Over much longer time span, reaching 5 Gyr, chaotic diffusion prevails, and we have performed an extensive statistical analysis of the orbital and rotational evolution of Mars, relying on Laskar's secular solution of the Solar System, based on more than 600 orbital and 200,000 obliquity solutions over 5 Gyr. The density functions of the eccentricity and obliquity are specified with simple analytical formulas. We found an averaged eccentricity of Mars over 5 Gyr of 0.0690 with standard deviation 0.0299, while the averaged value of the obliquity is 37.62° with a standard deviation of 13.82°, and a maximal value of 82.035°. We find that the probability for Mars' obliquity to have reached more than 60° in the past 1 Gyr is 63.0%, and 89.3% in 3 Gyr. Over 4 Gyr, the position of Mars' axis is given by a uniform distribution on a spherical cap limited by the obliquity 58.62°, with the addition of a random noise allowing a slow diffusion beyond this limit. We can also define a standard model of Mars' insolation parameters over 4 Gyr with the most probable values 0.068 for the eccentricity and 41.80° for the obliquity.

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*Keywords:* Mars, climate; Rotational dynamics; Resonances; Planetary dynamics

### 1. Introduction

The parameters of Mars' orbit and spin axis orientation control the global distribution and seasonal intensity of the solar insolation, and it is widely accepted that astronomical variations could have had a profound influence on its climatic history. These changes are probably characterized by a redistribution of the major martian volatiles (CO<sub>2</sub>, dust and water) and variations in their partition between atmospheric, surface, and subsurface reservoirs.

Because the permanent CO<sub>2</sub> cap is in equilibrium with the atmosphere, the martian atmospheric pressure is very sensitive to the polar temperature and thereby to obliquity. Climate models of CO<sub>2</sub> evolution over obliquity changes

which include exchange between atmospheric, polar caps, and regolith reservoirs suggest the possibility of large variations in atmospheric pressure (Ward et al., 1974; Toon et al., 1980; François et al., 1990; Fanale and Salvail, 1994). Depending on the total inventory of available CO<sub>2</sub>, occasional or prolonged periods of warmer, more clement, climate may have resulted throughout geological history (Jakosky et al., 1995).

In the same manner, the transport and redistribution of water between ground ice, surface ice, and atmospheric reservoirs appear to be largely sensitive to orbital parameters. Surface and near-surface ice stability is most strongly controlled by obliquity through the variations of the surface thermal forcing and of the abundance of water vapor sublimed in summer (e.g., Mellon and Jakosky, 1995; Jakosky et al., 1995). Simplified climate models suggest that during periods of high obliquity (> 40°), large quantities of polar ice could be sublimed and transported away

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to the tropical regions where it becomes stable (Jakosky and Carr, 1985; Jakosky et al., 1995). Such predictions were recently confirmed by full three-dimensional climate simulations (Haberle et al., 2000; Richardson and Wilson, 2000, 2002; Mischna et al., 2003; Levrard et al., 2003). Conversely, at present obliquities and lower, water ice is expected to be stable only in the high-latitude areas.

Many geological features provide the possibility of recent orbital-driven climatic changes. The most impressive is the extensive layering observed in the polar deposits and thought to contain alternate layers of water ice and dust in different proportions. Correlation between stratigraphic sequences and insolation parameters suggests that polar caps may preserve climatic records spanning the last few millions of years (Laskar et al., 2002). Additional morphological evidence may be found in Mars Global Surveyor observations of suspected recent water (gullies, paleolakes, outflow channels) or ice-generated (contraction-crack polygons, paleoglaciers) landforms (e.g., Baker, 2001; Mustard et al., 2001; Costard et al., 2002; Head and Marchant, 2003).

The presence of more ancient equatorial valley networks and putative oceanic shorelines, which support the occurrence of flowing water and different environmental conditions in early Mars (e.g., Baker et al., 1991), illustrates the extreme climate changes that Mars may have undergone throughout its geological history. In this context, long-term evolution and amplitudes of insolation parameters are a key element to understand the evolution of martian surface processes.

The history of the computation of the astronomical solution for the forcing of martian paleoclimates is very similar to what happened in the Earth's case (see Imbrie and Imbrie, 1979), but it occurred over a shorter time. The first climate models took only into consideration the precession of the axis of the planet, that alters the seasonal contrast (Leighton and Murray, 1966). Murray et al. (1973) then realized that the change of eccentricity of the planet resulting from secular planetary perturbations (from 0.004 to 0.141 in the solution of Brouwer and Van Woerkom (1950) that was then used), will modify significantly the insolation at the surface of the planet. The Earth spin axis obliquity undergoes variations of about  $\pm 1.3^\circ$  around its mean value ( $23.3^\circ$ ) (Laskar et al., 1993, and references therein) that are now recognized to have a large impact on the past climate of the Earth (see Imbrie and Imbrie, 1979). Ward (1973, 1974) was the first to realize that the obliquity of Mars suffers much larger variations, due to the proximity of secular spin orbit resonances. Using the orbital solution of Brouwer and Van Woerkom (1950) he found that the obliquity ( $\varepsilon$ ) of Mars was oscillating between  $\simeq 14.9^\circ$  and  $\simeq 35.5^\circ$ . This solution was later on improved, using the secular orbital solution of Bretagnon (1974), which led to slightly larger variations of the obliquity (Ward, 1979). A significant change arose with the use of the secular solution of Laskar (1988). Using this orbital solution, Ward and Rudy (1991) found that the obliquity evolution of Mars was critically depen-

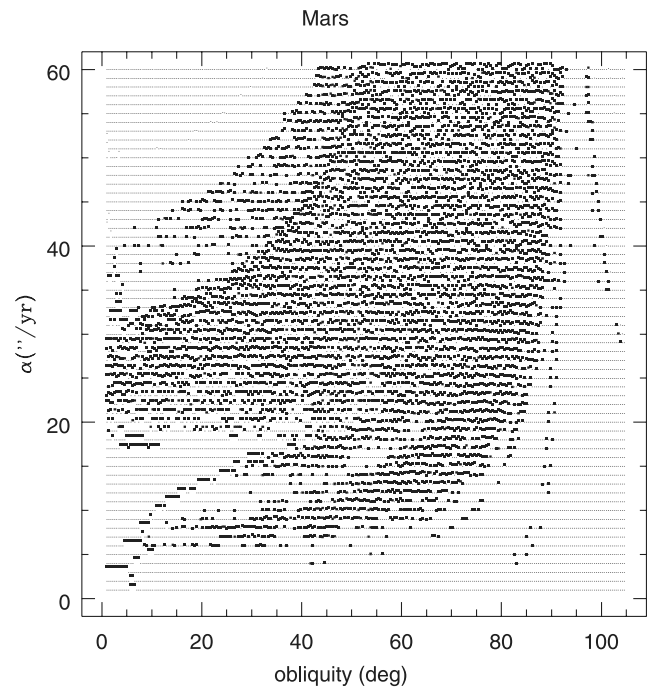


Fig. 1. Chaotic zone for the obliquity of Mars. The x-axis is the initial obliquity (in degrees) and the y-axis is the precession constant  $\alpha$ , in  $\text{arcsec yr}^{-1}$ . The regular solutions are represented by small dots, while large black dots denote the chaotic solutions (adapted from Laskar and Robutel, 1993).

dent on the initial precession rate, which was not known very precisely (see Section 3.2). The reason for this behavior was explained when it was demonstrated that the evolution of Mars' obliquity is chaotic (Laskar and Robutel, 1993; Touma and Wisdom, 1993). Moreover, (Laskar and Robutel, 1993) provided a global view for the dynamics of Mars' obliquity, describing the shape and extent of the associated chaotic zone (Figs. 1, 13). With this global portrait, and although the computations were only conducted over 45 Myr, they concluded that Mars' obliquity can wander from  $0^\circ$  to more than  $60^\circ$ .

Since then, the most significant improvement for the computation of an astronomical solution for martian paleoclimate studies is in the determination of the initial precession rate by the Pathfinder (Folkner et al., 1997) and Mars Global Surveyor space missions (Yoder et al., 2003), which now permit a reliable obliquity solution for Mars over a few million years.

After the pioneering work of Quinn et al. (1991) and Sussman and Wisdom (1992), and with the improvement of computer speed in the last decade, it becomes now possible to integrate numerically over hundreds of million of years (Myr) a dynamical model for the orbital evolution of the planetary orbits that is close to the ones used for short time ephemerides computations (Laskar, 2001; Laskar et al., 2004; Varadi et al., 2003). In the first section, we will present the derivation of the orbital solution for Mars over 250 Myr using the new symplectic integrator developed in our group (Laskar and Robutel, 2001). Particular care is taken to reduce the roundoff error, and the solution is com-

pared to the most up-to-date numerical ephemeris DE406 (Standish, 1998). The second part is devoted to the precession and obliquity equations and to the stability of the solutions with respect to the uncertainty of the parameters, and in particular of the initial precession rate. Because of the chaotic behavior of the obliquity, a precise solution cannot be derived over more than 10 to 20 Myr, but the equations are integrated over 250 Myr in order to establish statistics on the possible variations of the obliquity over this time span. In Section 5, this analysis is then continued over 5 Gyr. Over such an extended time, we do not use the direct numerical integration, but the secular equations of Laskar (1990), after some small adjustment of the parameters. This allows us to increase the computer speed by a factor of 2000 and to perform more extensive statistics, on both the orbital and obliquity solutions. We are then able to derive simple analytical expressions that fit extremely well with the density distribution of the eccentricity and obliquity over 5 Gyr (Section 5.4). In this sense, thanks to the chaotic behavior of the solutions, we can make very precise predictions on the obliquity and eccentricity evolution beyond 500 Myr, but these predictions are not for the exact values of these quantities, but for their probabilistic density functions.

## 2. Evolution over 250 Myr

In this first part, we will use a direct numerical integration of the planetary orbital motion in order to investigate the behavior of Mars' obliquity over 250 Myr. As the orbital motion is chaotic, even with a precise dynamical model, the computer roundoff numerical error alone will prevent obtaining a precise orbital solution for Mars over more than 60 Myr (see Fig. 4a). Moreover, the obliquity of Mars itself is chaotic, even more chaotic than its orbital motion (Laskar and Robutel, 1993; Touma and Wisdom, 1993). This will prevent even more drastically obtaining a precise solution for the obliquity over more than 10 to 20 Myr, with the present knowledge of the initial parameters (Section 3.2.1).

Our goal in this section will thus be to obtain a solution for the insolation parameters of Mars as precise as possible over 10 to 20 Myr for use in Mars paleoclimate studies. Then, with the same model, to explore the behavior of the solutions over 250 Myr and to derive a statistical vision of this chaotic system. Although no precise prediction is possible over this time interval, we will be able to derive a precise estimate of the density probability function for the evolution of the eccentricity of Mars and its obliquity. It is in fact paradoxical (see, for example, Lasota and Mackey, 1994) that it is actually the chaotic behavior of the system that will allow us to make a precise prediction of the evolution of the density function of Mars' orbital and rotational parameters.

### 2.1. Orbital motion

The orbital model comprises all 9 main planets of the Solar System, including Pluto. The post-Newtonian general

relativity corrections of order  $1/c^2$  due to the Sun are included following Saha and Tremaine (1994).

The Moon is treated as a separate object. In order to obtain a realistic evolution of the Earth–Moon system, we also take into account the most important coefficient ( $J_2$ ) in the gravitational potential of the Earth and of the Moon, and the tidal dissipation in the Earth–Moon system. We also integrate at the same time the precession and obliquity equations for the Earth and the evolution of its rotation period in a comprehensive and coherent way, following the lines of Néron de Surgy and Laskar (1997) and Correia et al. (2003). More details on the integration model can be found in (Laskar et al., 2004).

### 2.2. Numerical integrator

In order to minimize the accumulation of roundoff error, the numerical integration was performed with the new symplectic integrator scheme *SABAC*<sub>4</sub> of Laskar and Robutel (2001), with a correction step for the integration of the Moon. This integrator is particularly adapted to perturbed systems where the Hamiltonian governing the equations of motion can be written on the form  $H = A + \epsilon B$ , as the sum of an integrable part  $A$  (the Keplerian equations of the planets orbiting the Sun), and a small perturbation potential  $\epsilon B$  (here the small parameter  $\epsilon$  is of the order of the planetary masses). Using this integrator with step size  $\tau$  is then equivalent to integrating exactly a nearby Hamiltonian  $\tilde{H}$ , where the method's error  $H - \tilde{H}$  is of the order of  $O(\tau^8\epsilon) + O(\tau^2\epsilon^2)$ , and even  $O(\tau^8\epsilon) + O(\tau^4\epsilon^2)$  when the correction step is added, while the same quantity is of the order  $O(\tau^2\epsilon)$  in the widely used symplectic integrator of Wisdom and Holman (1991), and  $O(\tau^n\epsilon) + O(\tau^2\epsilon^2)$  with the correctors of Wisdom et al. (1996).

The step size used in most of the integration is  $\tau = 5 \times 10^{-3}$  yr = 1.82625 days. The initial conditions of the integration were least-square adjusted to the JPL ephemeris DE406 (Standish, 1998), in order to compensate for small differences in the model. In particular, we do not take into account the effect of the minor planets, and the modeling of the interactions in the Earth–Moon system is more complete in DE406 (see Williams et al., 2001).

In Fig. 2a is plotted the evolution of the total energy of the system from  $-250$  to  $+250$  Myr, after the removal of the secular trend that corresponds to the dissipation in the Earth–Moon system. The residuals are smaller than  $2.5 \times 10^{-10}$  after 250 Myr, and behave as a random walk with a standard deviation per step  $\sigma_1 \approx 2.7\epsilon_M$ , where  $\epsilon_M \approx 2.22 \times 10^{-16}$  is the machine Epsilon in double precision (Laskar et al., 2004). The normal component of the angular momentum is conserved over the same time with a relative error of less than  $1.3 \times 10^{-10}$  (Fig. 2b).

With these settings, the CPU time on a Compaq alpha workstation (833 MHz) is about 24 h for 5 Myr, and a 250 Myr run will last nearly two months.

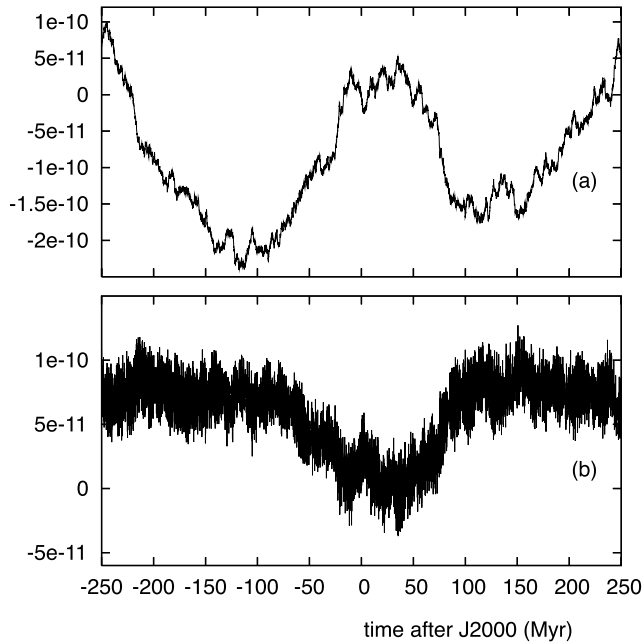


Fig. 2. Integrals conservation. (a) Relative variation of the total energy of the system versus time (in Myr) from  $-250$  to  $+250$  Myr (with origin at J2000) after correction of the secular trend due to the tidal dissipation in the Earth–Moon system. (b) Relative variation of the normal component of the total angular momentum.

### 2.3. Comparison with DE406

Using a direct numerical integrator, our goal is to provide a long term solution for the orbital and precessional elements of Mars with a precision that is comparable with the usual accuracy of a short time ephemeris.

We have thus compared our solution with the most advanced present numerical integration, DE406, that was itself adjusted to the observations (Standish, 1998). In the present paper, we will only discuss the orbital solution of Mars (Fig. 3).

Over the full range of DE406, that is from  $-5000$  to  $+1000$  yr from the present date, the maximum difference in Mars' longitude is less than  $0.28$  arcsec. These differences probably account in large part for the perturbations by the minor planets that are not taken into account in our computations. Over the whole interval, the difference in eccentricity is less than  $5 \times 10^{-8}$ , and less than  $0.08$  arcsec in inclination.

### 2.4. Variations on the orbital model

We will not present here all the wanderings that we had in the past years while searching for a precise model, which led us to obtain several intermediate solutions. A general discussion of the sensitivity of the orbital solution to the model, resulting from the chaotic behavior of the solution can be found in (Laskar, 1999a). Although this analysis was done with the secular equations, the conclusions would probably not be much changed using the complete equations.

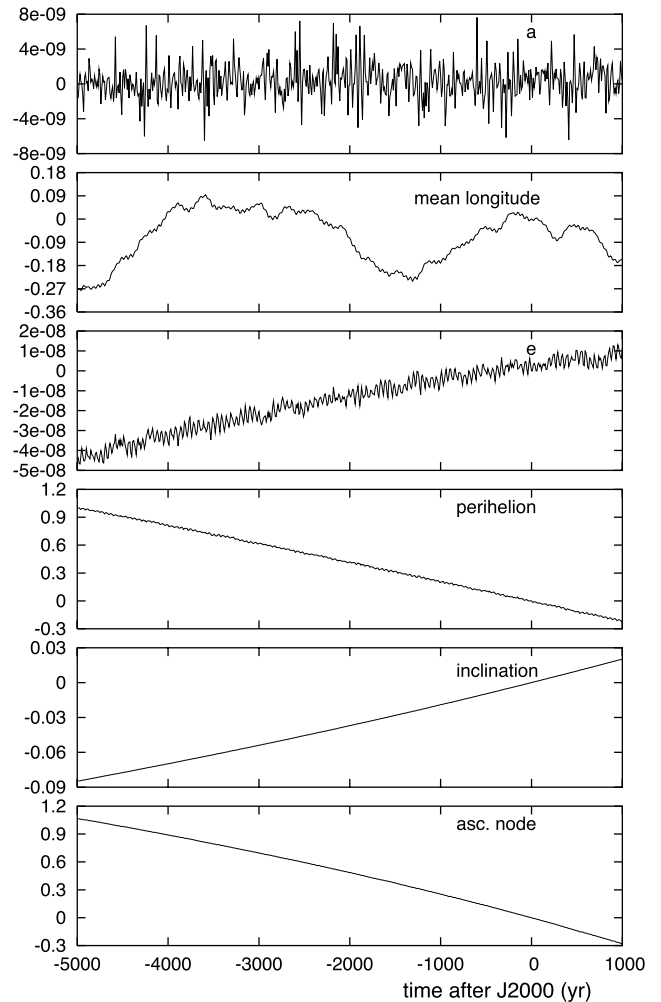


Fig. 3. Differences La2003 – DE406 over the full range of DE406 ( $-5000$  to  $+1000$  yr from J2000) for the orbital solution of Mars for all elliptical elements ( $a, \lambda, e, \varpi, i, \Omega$ ), respectively. The units for semi-major axis ( $a$ ) are AU, and arcsec for mean longitude ( $\lambda$ ), longitude of perihelion ( $\varpi$ ), inclination ( $i$ ), and longitude of the ascending node ( $\Omega$ ) from the ecliptic and equinox J2000.

The most recent determination of the solar oblateness ( $J_2$ ), obtained with the SOHO and GONG helioseismic data give  $J_2 = (2.18 \pm 0.06) \times 10^{-7}$  (Pijpers, 1998), with a very similar value adopted in DE406 ( $J_2 = 2 \times 10^{-7}$ , Standish, 1998), while in DE200, it was not taken into account (Newhall et al., 1983). Even with this small value, the absence of the solar oblateness in the dynamical model was identified in (Laskar, 1999a) as one of the main source of uncertainty in the long term solution for the Solar System. We will thus consider that comparing the nominal solution La2003 (with  $J_2 = 2 \times 10^{-7}$ ) with an alternate solution (La2003<sub>0</sub>) with  $J_2 = 0$  is representative of the uncertainty of the dynamical model for our long term integrations.

The results La2003 – La2003<sub>0</sub> for Mars' eccentricity are displayed in Fig. 4b over 100 Myr. The effect of the  $J_2$  becomes noticeable after about 30 Myr (26 Myr were predicted with an analytical estimate in (Laskar, 1999a)), and the solution remains very similar over 40 Myr, and totally out of

phase after 45 Myr. We will thus consider here that 40 Myr is about the time of validity of our present orbital solution for Mars.

In Fig. 4a, we have also tested the numerical stability of our numerical integration. This is done by comparison of the nominal solution La2003 (with stepsize  $\tau = 5 \times 10^{-3}$  yr) with an alternate solution, La2003\*, with the same dynamical model, and a very close stepsize  $\tau^* = 4.8828125 \times 10^{-3}$  yr. This special value was chosen in order that our output time span  $h = 1000$  yr corresponds to an integer number (204800) of steps, in order to avoid any interpolation problems in the check of the numerical accuracy. Figure 4a is thus a test of the time of validity for the obtention of a precise numerical solution with a given dynamical model, which is thus limited here to about 60 Myr.

This limitation of 60 Myr is at present a limitation for the time of validity of an orbital solution, independently of the precision of the dynamical model. In order to go beyond this limit, the only way will be to increase the numerical accuracy of our computations, by improving the numerical

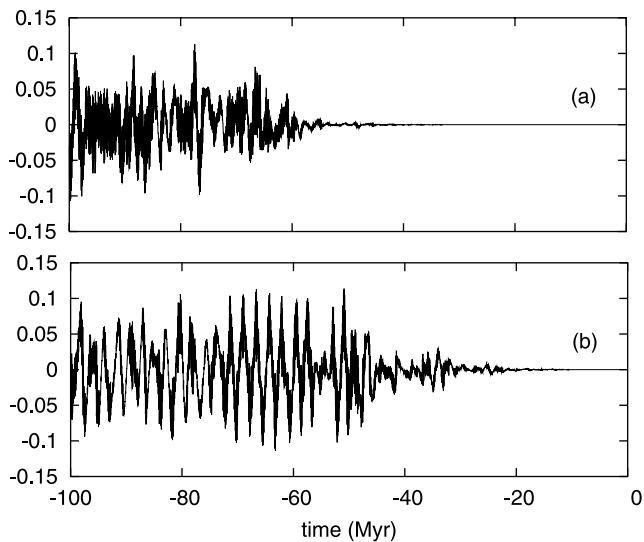


Fig. 4. Stability of the solution for Mars' eccentricity. (a) Difference of the nominal solution La2003 with stepsize  $\tau = 5 \times 10^{-3}$  yr, and La2003\*, obtained with  $\tau^* = 4.8828125 \times 10^{-3}$  yr. (b) Difference of the nominal solution with the solution obtained while setting  $J_2 = 0$  for the Sun (instead of  $2 \times 10^{-7}$  in the nominal solution).

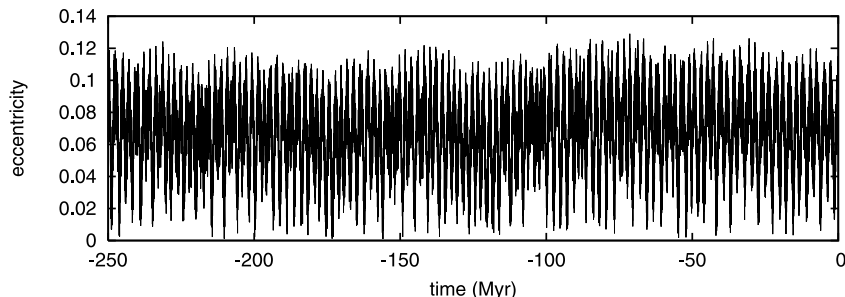


Fig. 5. The nominal solution La2003 for Mars' eccentricity over 250 Myr.

algorithm, or with an extended precision for the number representation in the computer. It should be noted that with our present algorithms, we are much more limited by the precision of the model (Fig. 4b) than by the numerical accuracy (Fig. 4a).

At this point, we need to stress that after showing that the solution is probably not precise over more than 40 Myr, and certainly not after 60 Myr, we are going in the remaining part of this paper to compute solutions over 250 Myr with the same algorithm (Fig. 5), and even 5 Gyr with the secular equations. This is indeed justified, as in this case we will not pretend to provide the actual solution for the evolution of the Solar System, but just investigate its possible behavior.

Finally, as it will be shown in the next sections, because of the strong chaos of the obliquity evolution of Mars, and of the uncertainties on its initial conditions, the obliquity solution will only be valid over 10 to 20 Myr, that is over much less time than the orbital computation. Over such a short time, the orbital present solution is certainly computed with a very good accuracy (Fig. 4b).

### 3. Precession equations

The equations for the evolution of precession and obliquity of Mars are rather simple, as contrary to the Earth's case, the effect of the satellites can be neglected. Indeed, it was shown by Goldreich (1965) that when a satellite is close to its planet, its orbit precesses about the planet's equatorial plane, as it is the case for Phobos and Deimos. In this case, the averaged torque exerted by the satellite is zero (see Laskar, 2004).

We suppose here that Mars is an homogeneous rigid body with moments of inertia  $A < B < C$  and we assume that its spin axis is also the principal axis of greatest inertia. The precession  $\psi$  and obliquity  $\varepsilon$  (Fig. 6) equations for a rigid planet in the presence of planetary perturbations are given by (Kinoshita, 1977; Laskar et al., 1993; Néron de Surgy and Laskar, 1997):

$$\begin{cases} \frac{d\varepsilon}{dt} = -\mathcal{B}(t) \sin \psi + \mathcal{A}(t) \cos \psi, \\ \frac{d\psi}{dt} = \alpha \cos \varepsilon - \cot \varepsilon (\mathcal{A}(t) \sin \psi + \mathcal{B}(t) \cos \psi) - 2\mathcal{C}(t), \end{cases} \quad (1)$$

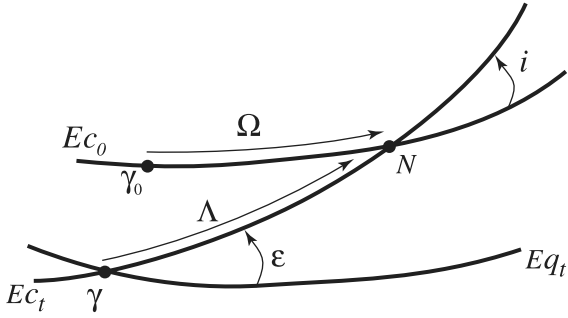


Fig. 6. Fundamental planes for the definition of precession and obliquity.  $Eq_t$  and  $Ec_t$  are the mean equator and ecliptic of Mars at date  $t$ .  $Ec_0$  is the fixed ecliptic of the Earth at Julian date J2000, with equinox  $\gamma_0$ . The general precession in longitude  $\psi$  is defined by  $\psi = \Lambda - \Omega$ .  $\Omega$  is the longitude of the node, and  $i$  the inclination. The angle  $\epsilon$  between  $Eq_t$  and  $Ec_t$  is the obliquity.

with

$$\begin{cases} \mathcal{A}(t) = \frac{2}{\sqrt{1-p^2-q^2}} [\dot{q} + p(q\dot{p} - p\dot{q})], \\ \mathcal{B}(t) = \frac{2}{\sqrt{1-p^2-q^2}} [\dot{p} - q(q\dot{p} - p\dot{q})], \\ \mathcal{C}(t) = q\dot{p} - p\dot{q}, \end{cases} \quad (2)$$

where  $q = \sin(i/2) \cos \Omega$  and  $p = \sin(i/2) \sin \Omega$ , and where  $\alpha$  is the “precession constant”:

$$\alpha = \frac{3G}{2\omega} \frac{m_\odot}{(a\sqrt{1-e^2})^3} E_d, \quad (3)$$

where  $m_\odot$  is the solar mass,  $G$  the gravitational constant,  $\omega$  the rotation rate of the planet, and  $E_d = (2C - A - B)/2C$  its dynamical ellipticity. For a fast rotating planet like Mars,  $E_d$  can be considered as proportional to  $\omega^2$ ; this corresponds to hydrostatic equilibrium (see, for example, Lambeck, 1980). In this approximation,  $\alpha$  is thus proportional to  $\omega$ . The quantities  $\mathcal{A}$ ,  $\mathcal{B}$ , and  $\mathcal{C}$  describe the driving terms due to the secular evolution of the orbital plane of the planet and are given by the integration of the planetary motions.

### 3.1. Dissipative effects

The possible dissipative effects influencing the evolution of Mars’ spin axis are: tidal dissipation due to the Sun, core–mantle friction, and climate friction.

#### 3.1.1. Tidal dissipation

We have considered the tidal contribution from the Sun, following the equations given in (Néron de Surgy and Laskar, 1997; Correia et al., 2003). Tidal contributions from the satellites Phobos and Deimos can be neglected compared with solar tides (Lambeck, 1979; Mignard, 1981). As shown in Fig. 7a, the effect of solar tidal dissipation is also very small, as it amounts to less than  $0.002^\circ$  after 10 Ma. As a consequence, the rotation rate of the planet did not change very much since the beginning of the Solar System. The choice of the tidal dissipation model is then not very relevant and we decided to use for Mars a constant- $Q$  model

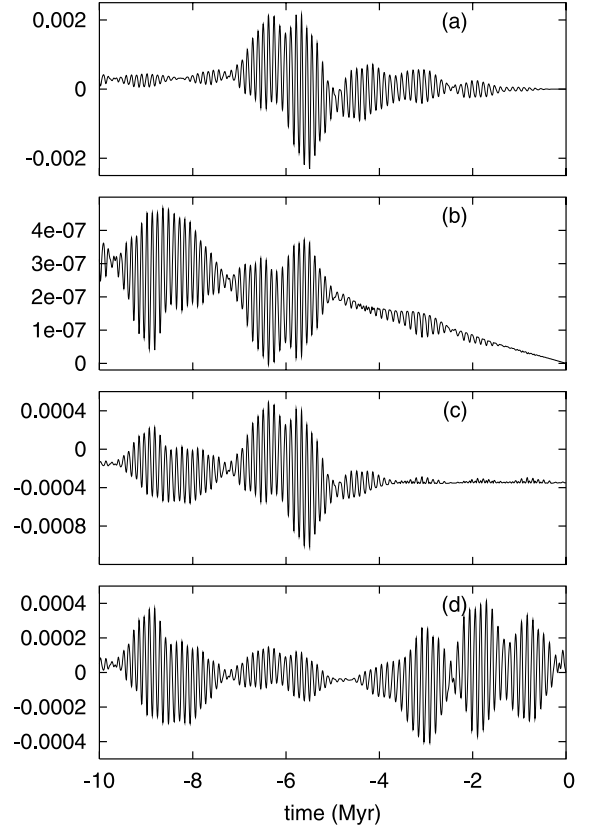


Fig. 7. Influence of parameter variation. The change (in degrees) observed over 10 Ma for the various parameters. (a) Effect of tidal dissipation; (b) core–mantle friction; (c)  $10 \times$  the uncertainty in the initial obliquity; (d)  $10 \times$  the uncertainty in the initial precession angle.

(see Correia et al., 2003, for the explicit equations). For this model, the dissipation is independent of the rotation rate, and its amplitude is proportional to  $k_2 \sin(2\omega\Delta t) \simeq k_2/Q$ , where  $\Delta t$  is the time delay between the tidal perturbation from the Sun and the consequent deformation of the planet. We used for the nominal solution La2003 a second Love number  $k_2 = 0.14$  and a dissipative factor  $Q = 50$ , while for the more recent La2004 solution (see Section 3.2.2) we used  $k_2 = 0.149$  and  $Q = 92$  from (Yoder et al., 2003). In both cases, the tidal contribution remains very small.

#### 3.1.2. Core–mantle friction

Little is known about the interior of Mars, since there are few geophysical constraints on its internal structure. However, the presence of a core can be inferred by the polar moment of inertia (Cook, 1977; Zhang, 1994). The existence of remnant crustal magnetism (Acuña et al., 1999) and the best determination of the polar moment of inertia so far (Folkner et al., 1997) imply that Mars’ core has substantial iron and that at least the outer part is liquid (Yoder et al., 2003). Because of their different densities, the core and the mantle do not have the same dynamical ellipticity. Since the precession torques exerted by the Sun on Mars’ core and mantle are proportional to this quantity, the two parts tend to precess differently around an axis perpendicular to the or-

bital plane (this results from Poincaré’s study (1910) on the motion of an inviscid fluid contained in a rotating ellipsoidal shell). This tendency is more or less counteracted essentially by two different interactions produced at the interface: the torque of non-radial inertial pressure forces of the mantle over the core provoked by the non-spherical shape of their interface; and the torque of the viscous friction between the core and the mantle (e.g., Correia et al., 2003). The effect of core viscosity in the Earth’s case was treated by Stewartson and Roberts (1963) and Roberts and Stewartson (1965) for low values of viscosity, by linearizing the equations for the viscous boundary layer. Busse (1968) further studied the effect of the non-linear advective term in the equations. The contribution of core–mantle friction to the secular variation of the obliquity is given by (Rochester, 1976; Pais et al., 1999):

$$\dot{\varepsilon} \simeq -\kappa \frac{\alpha^2 \cos^3 \varepsilon \sin \varepsilon}{\gamma_{\text{el}} C E_c^2 \omega^2}, \quad (4)$$

with (Roberts and Stewartson, 1965; Busse, 1968)

$$\kappa \simeq 2.62 \frac{C_c \sqrt{\omega \nu}}{R_c}, \quad (5)$$

where  $C_c$ ,  $E_c$ , and  $R_c$  are respectively the core’s polar moment of inertia, dynamical ellipticity and radius;  $\nu$  is the kinematic viscosity and  $\gamma_{\text{el}} \simeq 0.75$  a correcting factor accounting for the elastic deformation of the mantle. The kinematic viscosity ( $\nu$ ) is poorly known. Even in the case of the Earth, its uncertainty covers about 13 orders of magnitude (Lumb and Aldridge, 1991). It can be as small as  $\nu = 10^{-7} \text{ m}^2 \text{ s}^{-1}$  for the Maxwellian relaxation time and experimental values for liquid metals or as big as  $\nu = 10^5 \text{ m}^2 \text{ s}^{-1}$  for the damping of the Chandler wobble or attenuation of shear waves. The best estimate so far of the actual value of this parameter is  $\nu \simeq 10^{-6} \text{ m}^2 \text{ s}^{-1}$  (Gans, 1972; Poirier, 1988). The secular obliquity variations given by expression (4) are of the same order as the tidal variations for the highest values of the viscosity and can be neglected for the best estimations. Indeed, with  $Q = 100$  we compute for tidal evolution:

$$\dot{\varepsilon}_{\text{tides}} \sim 10^{-7} \text{ arcsec yr}^{-1}. \quad (6)$$

Using  $R_c = 1.68 \times 10^6 \text{ m}$  (Yoder et al., 2003),  $C_c = 0.06C$  (we compute  $C_c = 8\pi \rho_{\text{Fe}} R_c^5 / 15$ , with  $\rho_{\text{Fe}} \simeq 7 \text{ g cm}^{-3}$ ) and  $E_c = 0.5E_d$  (we compute  $E_c/E_d = C R_c^5 / C_c R^5$ ), we have for secular obliquity variations resulting from core–mantle friction:

$$\dot{\varepsilon}_{\text{cmf}} \sim 10^{-7} \sqrt{\nu_{[\text{m}^2 \text{ s}^{-1}]}} \text{ arcsec yr}^{-1}. \quad (7)$$

For weak viscosity like the best estimations for the Earth ( $\nu \sim 10^{-6} \text{ m}^2 \text{ s}^{-1}$ ), we have  $\dot{\varepsilon}_{\text{cmf}} \sim 10^{-3} \dot{\varepsilon}_{\text{tides}}$ , that is, the core–mantle friction effect can be neglected with respect to solar tides (Fig. 7b). Since this parameter is very uncertain, we can expect that electromagnetic coupling or turbulence to increase friction at the interface between the core and the mantle. We can then talk about an effective viscosity

some orders of magnitude higher, but we see that even for  $\nu \sim 1 \text{ m}^2 \text{ s}^{-1}$ , we still have  $\dot{\varepsilon}_{\text{cmf}} \sim \dot{\varepsilon}_{\text{tides}}$ . The core–mantle friction effect is then of the same order of tidal effects or even weaker.

### 3.1.3. Climate friction

Climate friction is a positive feedback process between obliquity variations and the resulting redistribution of volatiles at the planetary surface that affect its dynamical ellipticity. Although a significant fraction of the variations of surface loading is compensated by the visco-elastic adjustment of the internal planetary mass, delayed responses in climatic and viscous relaxation processes may lead to a secular term in the obliquity evolution. Conversely to previous dissipative effects, climate friction not only depends on the instantaneous value of the obliquity but also on its dynamical evolution.

The impact of CO<sub>2</sub> and water caps has been respectively studied by Rubincam (1990, 1993) and Bills (1999) using linear approximations for the obliquity dynamics and global mass redistribution. Because both the volatile response to obliquity forcing and martian internal parameters (density, elasticity, rigidity, viscosity) are still poorly constrained, long-term estimation of the climate friction impact is very uncertain. Similar analyses suggest that obliquity–oblateness feedback has probably not changed the Earth’s obliquity by more than 0.01° per Myr during the severe recent Pliocene–Pleistocene glaciations ( $\sim 0\text{--}3 \text{ Ma}$ ) (Levrard and Laskar, 2003). Since the martian caps are about one order of magnitude less massive than the water/ice mass involved in typical terrestrial ice age, we expect that its impact is negligible over the last 20 Myr, compared to other sources of uncertainty, and it was not taken into account in our long-term obliquity solutions.

Other geophysical processes, such as volcanic events and uplifts, construction of the large Tharsis province or Hellas impact basins, or formation of mass anomalies associated with mantle convection could have produced changes of early Mars’ obliquity (Ward et al., 1979), but they do not affect the recent 20 Myr of martian history.

## 3.2. Initial spin axis orientation

As the motion of Mars’ spin axis is chaotic (Laskar and Robutel, 1993; Touma and Wisdom, 1993), its evolution critically depends on its precise initial conditions. Significant improvements were made on this determination with the results of the Pathfinder mission (Folkner et al., 1997) which provided some estimates on the determinations of the initial conditions for the spin of Mars (Table 1). According to their paper, this uncertainty represents five times the standard error ( $\sigma$ ) obtained during their fitting process to the data. This is more than what is usually done ( $3\sigma$ ), but according to the authors, this allows for ‘the failure to account for various systematic effects.’ We will be thus even more cautious on these data, and investigate its possible variations up to  $10\sigma$ ,

Table 1  
Initial conditions from (Folkner et al., 1997)

Parameter	Value
Obliquity $\varepsilon$ (degrees)	25.189417 (35)
Node $\psi$ (degrees)	35.43777 (14)
Precession rate $d\psi/dt$ (mas yr <sup>-1</sup> )	-7576 (35)
Rotation rate $\omega$ (degrees day <sup>-1</sup> )	350.89198521 (8)

The uncertainty is given in parentheses in the units of the last displayed digit.

that is  $2\delta p$ , around  $p_0$  where

$$p_0 = -7.576 \text{ arcsec yr}^{-1},$$

$$\delta p = 0.035 \text{ arcsec yr}^{-1} \quad (8)$$

are the initial value of the precession rate and its uncertainty given in Table 1.

First of all, we will show that the uncertainty on the initial position of the pole axis is sufficiently small and does not induce significant changes in the solution. We have thus run some numerical integrations with a deviation of 10 times this uncertainty in both the initial obliquity (Fig. 7c) and precession angle (Fig. 7d). In both cases, the maximum difference with respect to the nominal solution is less than  $0.001^\circ$  after 10 Myr, and thus negligible, compared to other uncertainties.

### 3.2.1. Precession constant

In fact, the main source of uncertainty in the long term evolution of the obliquity of Mars is the uncertainty in the initial precession rate of Mars. Nevertheless, even when being pessimistic with respect to the accuracy of the published determination (Table 1), the situation after the Pathfinder mission is much improved, and it is now possible to derive results over 10 Myr with confidence. In order to test the stability of the obliquity solution with respect to the uncertainty  $\delta p$  of the initial precession frequency  $p$ , we have integrated 401 different solutions over 100 Myr, using always as an input the same orbital solution, and with an initial precession frequency  $p_k = p_0 + k \times \delta p/100$ , for  $k = -200, \dots, 200$ . The different solutions are very similar in the vicinity of the origin, but after a few millions of years, they diverge significantly, as a result of the chaotic behavior of the spin precession, resulting from the secular perturbations of the other planets (Laskar and Robutel, 1993).

In Fig. 8b, we have summarized the obliquity evolution of all these solutions. For each solution  $\varepsilon_k(t)$ , ( $k = -200, \dots, 200$ ), we have computed the maximum  $M_k(t)$  and minimal value  $m_k(t)$  reached by the obliquity over 1 Myr. We have then taken the upper and lower envelopes of these curves. The 4 curves are thus (in decreasing value at 100 Myr)  $S_1 = \text{Max}(M_k)$ ,  $S_2 = \text{Max}(m_k)$ ,  $S_3 = \text{Min}(M_k)$ ,  $S_4 = \text{Min}(m_k)$ . They represent the extreme variations of the obliquity over 100 Myr. As was first shown in (Laskar and Robutel, 1993), the variation of the obliquity ranges from  $0^\circ$  to more than  $60^\circ$ .

The differences of the curves  $S_1, S_3$  and  $S_2, S_4$  represent the variability of the solutions. It is thus clear that if the un-

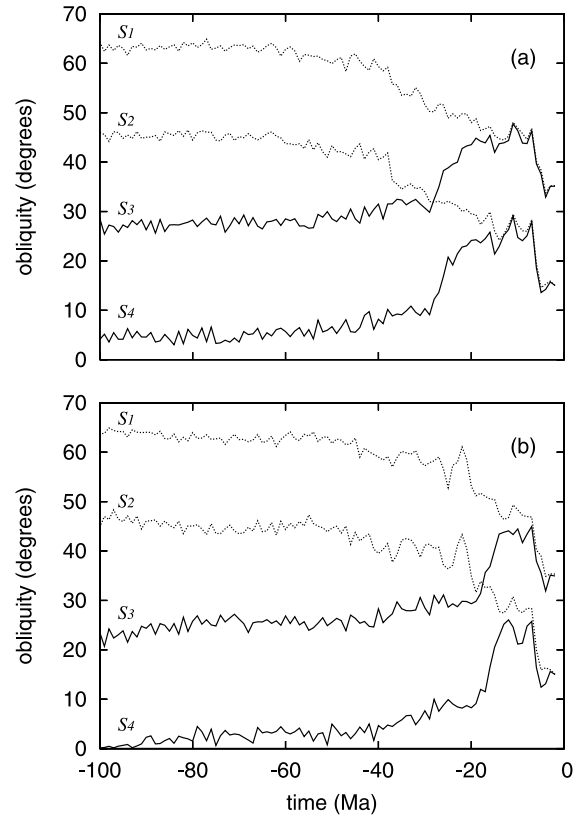


Fig. 8. Variations in the obliquity solution due to the uncertainty in the precession frequency. In (a), 1001 solutions of the obliquity are computed, with initial precession values ranging from  $p_0 - \delta p/2$  to  $p_0 + \delta p/2$  with an even stepsize. In (b) the same analysis is made with 401 solutions with initial precession values ranging from  $p_0 - 2\delta p$  to  $p_0 + 2\delta p$ . The maximum  $M_k$  and minimum  $m_k$  envelopes over 1 Myr are computed for each solution  $k$ . The plotted curves are then:  $S_1 = \text{max}(M_k)$ ,  $S_2 = \text{max}(m_k)$ ,  $S_3 = \text{min}(M_k)$ ,  $S_4 = \text{min}(m_k)$ .

certainty on the precession frequency  $p$  is  $2\delta p$ , the solution for the obliquity cannot be precise over more than 10 Myr.

We have repeated the same experience with a more optimistic view, with an initial precession frequency  $p_k = p_0 + k \times \delta p/1000$ , for  $k = -500, \dots, 500$ , that is on the interval  $p_0 - \delta p/2, p_0 + \delta p/2$ . In this case, the variation of the obliquity also ranges from  $0^\circ$  to more than  $60^\circ$ , but the solution appears to be valid over nearly 20 Myr (Fig. 8a).

Depending on the reliability of the uncertainty on  $p_0$  given in Table 1, we can thus provide a reliable solution for the obliquity of Mars over 10 to 20 Myr. The nominal solution (denoted La2003) can be retrieved from the Web site <http://www.imcce.fr/Equipes/ASD/mars.html> from  $-20$  to  $+10$  Myr (Fig. 9) together with the subroutine for the computation of the various insolation quantities following Laskar et al. (1993). This is the solution that was used for the paleoclimate analysis of Mars polar cap layers in (Laskar et al., 2002). A very important constraint in this analysis was the observation that despite the uncertainty of the precession constant, and even using  $2\delta p$  for this uncertainty, all the solutions presented a large increase of the obliquity at 5 Myr (Figs. 8b, 10), already observed in some solutions of Ward



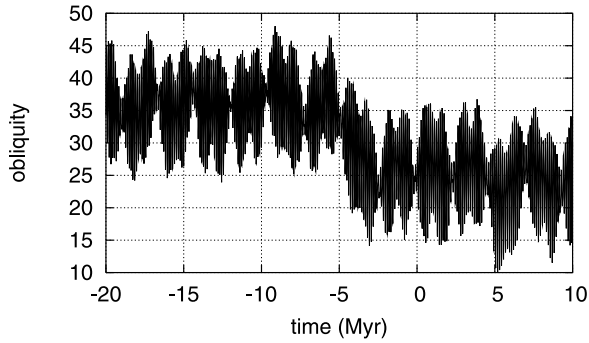


Fig. 9. Obliquity (in degrees) of the nominal solution La2003 from  $-20$  to  $+10$  Myr.

and Rudy (1991) and Touma and Wisdom (1993), whose effect should have been noticeable on the ice record.

### 3.2.2. The solution La2004

The previous discussion demonstrates the importance of a better determination of Mars' initial precession rate, as this will be the only way to extend further our knowledge of the past insolation on the planet. We will thus be attentive to the next space missions to Mars with geodesic studies. The Netlander mission, initially projected in 2005, is a very good candidate for this goal (Barriot et al., 2001). One should consider that the determination with only the Viking mission gave  $p = -7.83 \pm 0.3 \text{ arcsec yr}^{-1}$  (probably  $3\sigma$ ) (Yoder and Standish, 1997), while the addition of the Pathfinder mission gave  $p = -7.576 \pm 0.035 \text{ arcsec yr}^{-1}$  ( $5\sigma$ ). More recently, using the Mars Global Surveyor data, Yoder et al. (2003) gave a new determination of the precession constant as  $p = -7.597 \pm 0.025 \text{ arcsec yr}^{-1}$ . The uncertainty is still  $5\sigma$ , and is slightly improved with respect to the previous determination. We have thus updated our solution using these new data. Moreover Yoder et al. (2003) give also an improved determination of the tidal coefficients  $k_2 = 0.149 \pm 0.017$  and  $Q = 92 \pm 11$  that are different from the values of our nominal solution ( $k_2 = 0.14$  and  $Q = 50$ ). As the tidal dissipation is very small (Fig. 7), this has no noticeable effect on the solution, but we have updated these data to their new values as well. The resulting solution will thus be called La2004, and will be provided as well on our Website <http://www.imcce.fr/Equipes/ASD/mars.html>.

The difference of the two solutions over 20 Myr is small (compare Figs. 9 and 10a), and we have not done again the previous stability analysis with the updated values of La2004. The solution La2004 for the obliquity, eccentricity, and insolation in summer at the North pole is given in Fig. 10.

## 4. Variations over 250 Ma

We have plotted in Fig. 11 several examples of the solutions obtained by changing the initial precession rate  $p$  by a small amount. In this figure, the label of each panel indicates

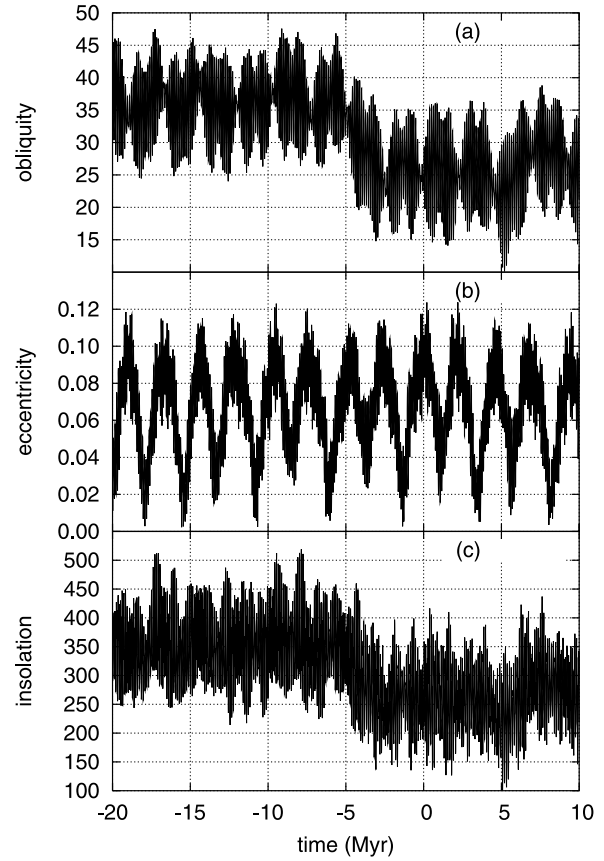


Fig. 10. Obliquity (in degrees) (a), eccentricity (b), and insolation (c) (in  $\text{watt m}^{-2}$ ) at the north pole surface at the summer solstice ( $L_S = 90^\circ$ ) for the solution La2004 from  $-20$  to  $+10$  Myr.

the offset in  $p$  in units of  $\delta p/100$ . In the labels, “Pxxx” corresponds to an initial precession rate  $p = p_0 + \text{xxx} \times \delta p/1000$ , and “Nxxx” to  $p = p_0 - \text{xxx} \times \delta p/1000$ . The solution “301003BIN\_A.P000” is thus the nominal solution La2003. All the solutions in this figure are thus within  $p_0 \pm \delta p/5$  and can be considered as equiprobable. They represent the variety of all the computed 1001 solutions, and some typical behavior.

In particular, we put in this plot the two solutions with closest initial conditions  $p_0 \pm \delta p/1000$  from the nominal solution, and one can thus very well understand, by looking at the very different behavior of the three solutions “301003BIN\_A.N001,” “301003BIN\_A.P000,” “301003BIN\_A.P001,” that it will be very difficult to predict precisely the past evolution of Mars' obliquity over more than a few tens of millions of years. We are thus left with the only possibility; that is, to search for the possible behavior of the obliquity in the past. We will first look to the maximum possible variations of the obliquity, and then to its most probable evolution, by computing a density function for the main factors of the climate variation, the obliquity and the eccentricity. The longitude of perihelion from the moving equinox is also important for the determination of the insolation on the surface of the planet, but after a few millions

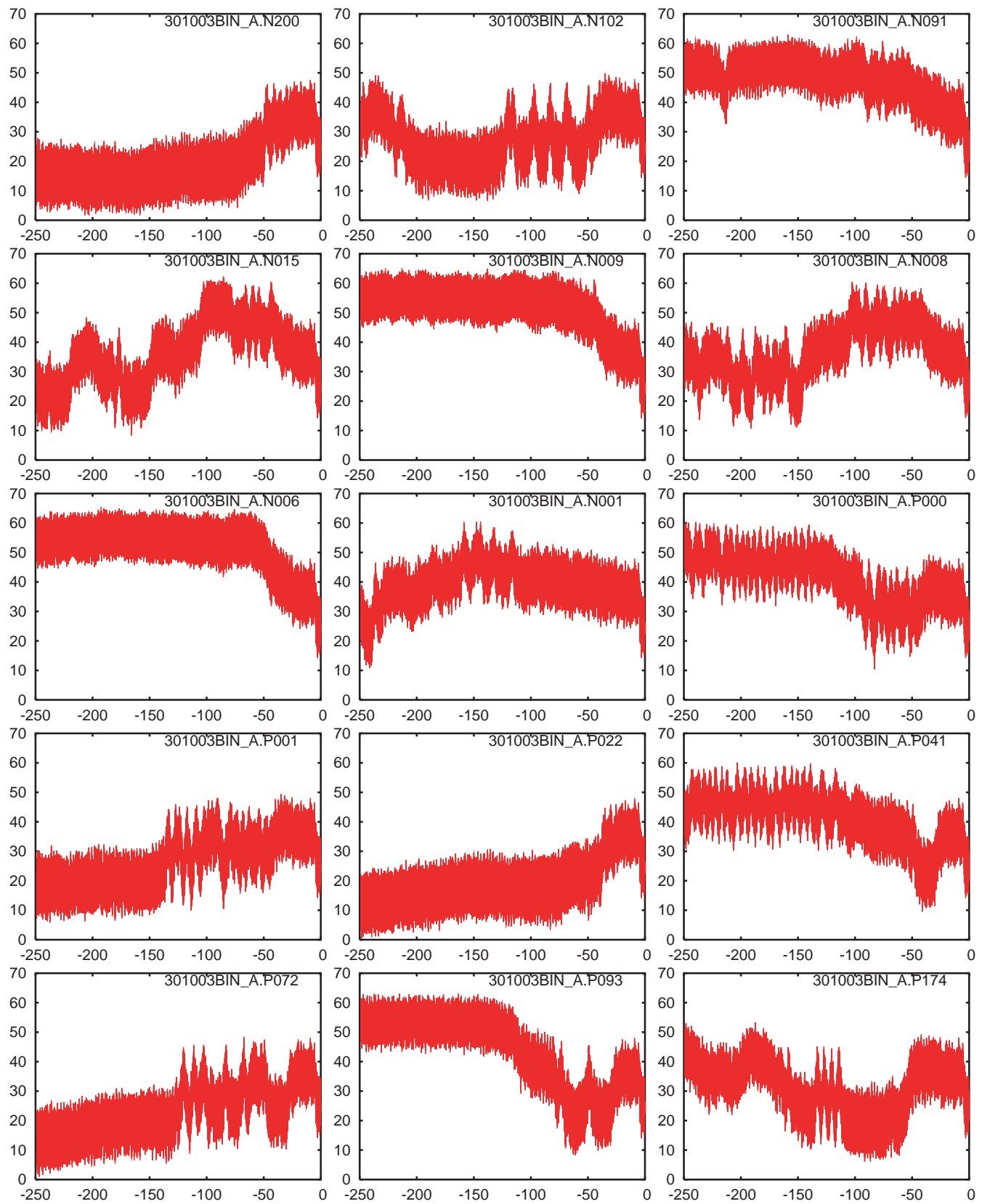


Fig. 11. Examples of possible evolution of Mars' obliquity over the past 250 Myr.

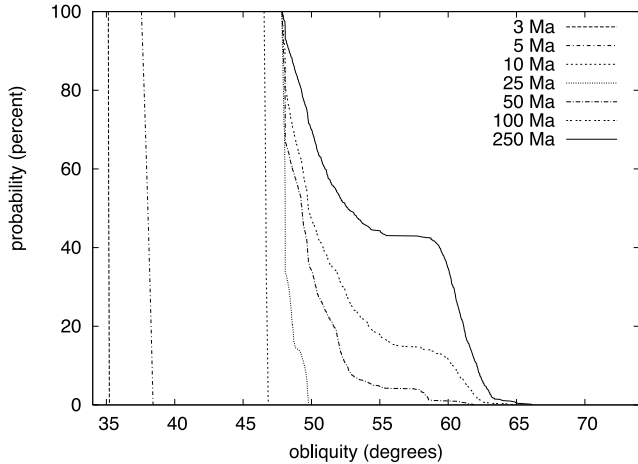


Fig. 12. The probability (vertical axis) of reaching a given obliquity (horizontal axis) after a specified time is given by the different curves, for 3, 5, 10, 25, 50, 100, and 250 Myr. The estimate is obtained by 1001 integrations for regularly spaced values of  $p$  ranging from  $p_0 - \delta p/2$  arcsec yr<sup>-1</sup> to  $p_0 + \delta p/2$  (Eq. (8)).

of years, we can assume that it will be entirely randomized over  $[0, 2\pi]$ .

#### 4.1. Maximum variations over 250 Ma

As we cannot display the 1001 examples that we computed, with  $p \in [p_0 - \delta p/2, p_0 + \delta p/2]$ , for the possible obliquity evolution over 250 Myr, we have summarized all the different cases by computing the maximum obliquity reached by the different solutions over a given prescribed time. The results are then transcribed in terms of probability to reach a given obliquity before a given time (Fig. 12).

For 3, 5, or 10 Myr, the plotted curve is nearly a vertical line. This reflects the fact that, over 10 Myr, all the solutions are very close to each other, and behave in the same way. So they all reach 35° before 3 Myr, remain below 40° until 5 Myr, and then reach about 48° before 10 Myr.

After 10 Myr, the situation is different, as due to the chaotic diffusion of the trajectories, some of them will reach high values of the obliquity. We observe that over 25 Myr, the effect of chaos is already visible, and some of the orbits reach 50°. After 100 Myr, a significant part of the orbits (about 13%) went beyond 60°.

#### 4.2. The chaotic zone

In Fig. 12, it appears that when the obliquity reaches 55°, then it rapidly reaches 60° as well. This can be understood by looking to the shape of the chaotic zone for Mars obliquity (Fig. 13). This plot (adapted from Laskar and Robutel, 1993, and Laskar, 1996) is obtained using Frequency Map analysis (Laskar, 1990, 1999b). In Fig. 13a, a refined determination of the precession frequency  $p$  (in arcsec yr<sup>-1</sup>) is plotted versus the initial obliquity (in degrees). It can be shown (Laskar, 1999b) that regular trajectories will corre-

spond to a smooth part of this curve, while chaotic regions will occur when the frequency curve is not regular.

The chaotic zone is divided essentially in two boxes  $B1, B2$ , that appear clearly on Figs. 13a, 13b. The first box  $B2$  ranges from about 10° to 50°, while  $B1$  extends from about 30° to 60° (the actual size of these boxes can be measured on the ordinate axis of Fig. 13b).  $B1$  mostly results from the secular perturbations related to the  $s_1$  secular proper mode related to Mercury, while  $B2$  is mostly due to  $s_2$  that is related to Venus. Each of these two modes  $s_1, s_2$  is associated to multiple side terms (Fig. 13c) resulting from the complex behavior of the orbital motion of the inner Solar System, and in particular from the presence of the secular resonance associated with  $(g_1 - g_5) - (s_1 - s_2)$  (see Laskar, 1990, for more details). As a consequence for the obliquity, all the resonances with these side terms will overlap, and thus increase the chaotic zone around  $s_1$  and  $s_2$ . Inside  $B1$  or  $B2$ , the chaotic diffusion is large, and the zone is completely explored in a relatively short time, but the communication between the two boxes is more difficult. Occasionally, a trajectory initially in  $B2$  will find the door to  $B1$ , and its obliquity will reach values larger than 50°. But as it is now in the  $B1$  box, the strong chaotic diffusion of this box will lead the solution very rapidly to more than 60°. Beyond 60°, the frequency curve is more regular, but still not completely smooth. We can thus expect some slow diffusion to higher values of the obliquity. Moreover, the frequency map of Fig. 13a is obtained over a relatively short time. Over longer times the slow diffusion of the orbital solution itself will change slightly the values of the main frequencies (Laskar, 1990), and the chaotic zones will be displaced, thus sweeping a larger area.<sup>1</sup>

#### 4.3. Obliquity statistics over 250 Myr

The maximum value reached by the obliquity (Fig. 12) is an important quantity, as even if these extreme situations (like  $\varepsilon > 60^\circ$ ) may not last for a long time, they could induce important effects on the martian climate evolution. But now we will look for the most probable situation, by computing the density distribution of the obliquity over various time spans for all our 1001 solutions. For this purpose, we have sampled all these solutions with a sampling rate of 5000 yr, and plotted the normalized density function  $f_T(\varepsilon)$  over successive time spans  $T = 3, 50, 100, 250$  Myr (Fig. 14).

It is striking to see that over relatively short time interval, as 3 Myr, when the chaotic behavior is not noticeable, the distribution is very irregular, but as the time interval  $T$  increases, the density function  $f_T(\varepsilon)$  becomes very smooth

<sup>1</sup> In their integrations, Touma and Wisdom (1993) found that Mars' obliquity only increases up to 49°. This can be easily understood with the help of Fig. 13. Most probably, the solutions explored by Touma and Wisdom remained only in  $B2$  during their limited integration time.

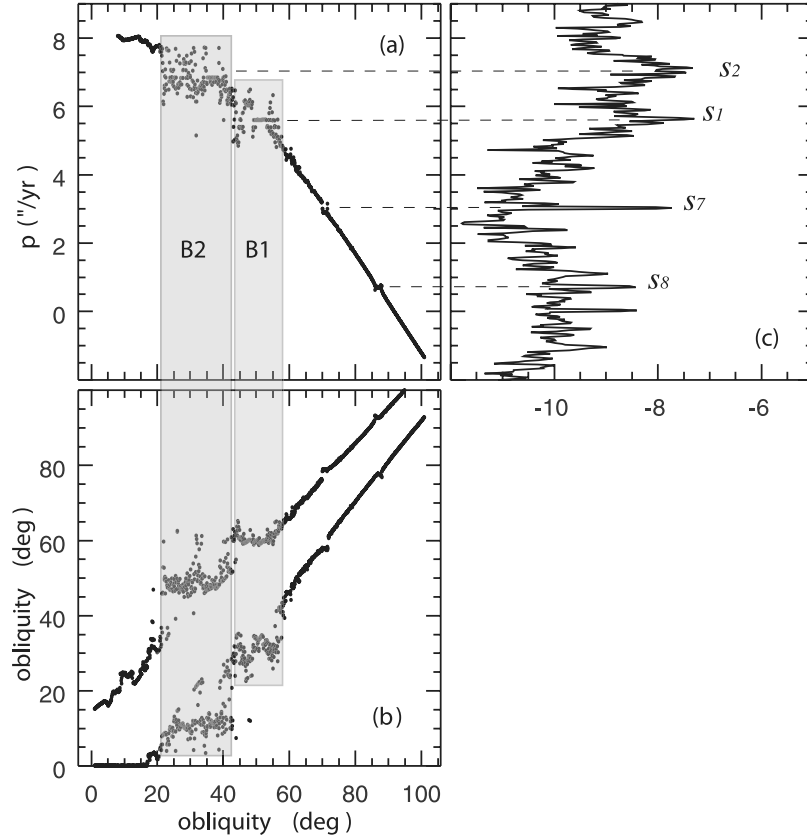


Fig. 13. Frequency analysis of Mars' obliquity. (a) The frequency map is obtained by reporting in the ordinate the value of the precession frequency obtained for each of the 1000 integrations over 56 Myr for the different values of the initial obliquity (abscissa). A large chaotic zone is visible, ranging from  $0^\circ$  to about  $60^\circ$ , with two distinct zones of large chaos, B1 and B2. (b) Maximum and minimum values of the obliquity reached over 56 Myr. In (c), the power spectrum of the orbital forcing term  $\mathcal{A} + i\mathcal{B}$  (Eq. (2)) is given in logarithmic scale, showing the correspondence of the chaotic zone with the main secular frequencies  $s_1, s_2, s_7, s_8$  (adapted from Laskar and Robutel, 1993, and Laskar, 1996).

and very close to a pure Gaussian density  $G_{m,\sigma}(x)$

$$G_{m,\sigma}(x) = \frac{1}{\sqrt{2\pi}\sigma} \exp\left(-\frac{(x-m)^2}{2\sigma^2}\right) \quad (9)$$

with mean value  $m$  and standard deviation  $\sigma$ . Indeed, in Fig. 14, the obliquity distribution is plotted in full line, while we plotted in dashed line the Gaussian distribution  $G_{m,\sigma}(x)$  with mean  $m = \bar{\varepsilon}$  and standard deviation  $\sigma = \sigma_\varepsilon$ . The corresponding values for  $\bar{\varepsilon}$  and  $\sigma_\varepsilon$  over the various values of  $T$  are given in Table 2.

## 5. Diffusion over 5 Gyr

Now, we will analyze the possible evolution of the orbital and precessional solution over 5 Gyr, that is, over a time comparable to the age of the Solar System. Over this time span, we will be able better to understand the effect of the slow chaotic diffusion of the orbital motion of the Solar System.

Indeed, when the main secular frequencies of the orbital motion will change, as a result of the chaotic diffusion, the chaotic region in the frequency space (Fig. 13) will be modified, and the obliquity may enter more easily in alternate

Table 2  
Statistics for the obliquity evolution over 250 Myr

$T$ (Myr)	$\varepsilon_M$ ( $^\circ$ )	$\varepsilon_m$ ( $^\circ$ )	$\bar{\varepsilon}$ ( $^\circ$ )	$\sigma_\varepsilon$ ( $^\circ$ )
3	35.218	15.009	25.254	4.992
5	38.411	13.578	26.874	5.845
10	46.859	13.578	31.468	7.187
25	49.781	13.578	34.218	6.543
50	61.903	5.289	34.427	7.099
100	64.898	3.014	34.199	8.354
250	66.154	0.043	34.643	9.966

Maximum ( $\varepsilon_M$ ), minimum ( $\varepsilon_m$ ), average ( $\bar{\varepsilon}$ ), and standard deviation ( $\sigma_\varepsilon$ ) of the obliquity are given over different time intervals (col. 1). The nominal orbital solution is used, and the statistics are performed over 1001 obliquity solutions with initial conditions for regularly spaced values of  $p$  ranging from  $p_0 - \delta p/2$  to  $p_0 + \delta p/2$ .

regions of the phase space. We thus expect that the diffusion of the planetary trajectories will lead to a larger diffusion of the obliquity.

### 5.1. The secular equations

In order to investigate the diffusion of the orbits over 5 Gyr, we will use the secular equations of Laskar (1990), with some small modifications. The secular equations are

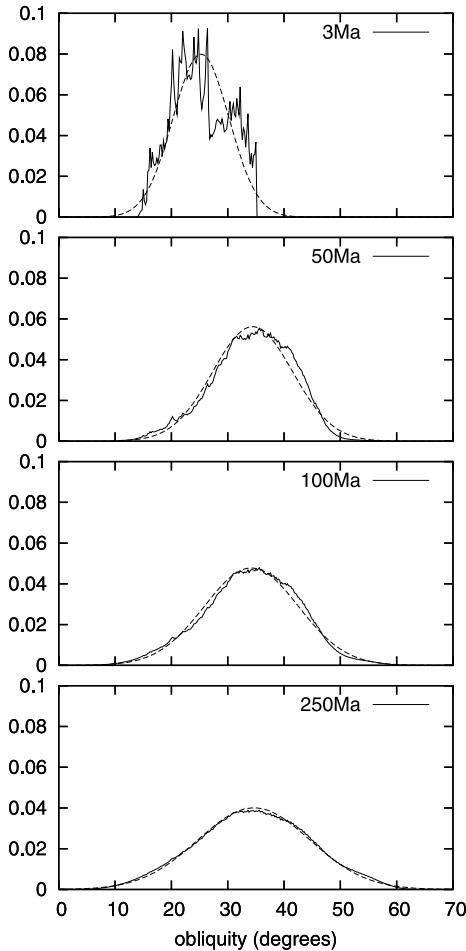


Fig. 14. Normalized density function for the obliquity of Mars over 3, 50, 100, and 250 Myr (full lines). The dashed lines are the best fit with a Gaussian density.

obtained by averaging the equations of motion over the fast angles that are the mean longitudes of the planets. Indeed, due to the degeneracy of the Keplerian problem, the other coordinate angles (longitudes of the perihelion and node) evolve much more slowly, as their motion results uniquely from the mutual planetary perturbations, and the perturbation of the pure Newtonian point mass attraction of the Sun from general relativity and quadrupole moment of the Sun. The averaging of the equation of motion is obtained by expanding the perturbations of the Keplerian orbits in Fourier series of the angles, where the coefficients themselves are expanded in series of the eccentricities and inclinations. This averaging process was conducted in a very extensive way, up to second order with respect to the masses, and through degree 5 in eccentricity and inclination, leading to truncated secular equations of the Solar System of the form

$$\frac{dw}{dt} = \sqrt{-1} \{ \Gamma w + \Phi_3(w, \bar{w}) + \Phi_5(w, \bar{w}) \}, \quad (10)$$

where  $w = (z_1, \dots, z_8, \zeta_1, \dots, \zeta_8)$ , with  $z_k = e_k \exp(i\varpi_k)$ ,  $\zeta_k = \sin(i_k/2) \exp(i\Omega_k)$  ( $\varpi_k$  is the longitude of the perihelion). The  $16 \times 16$  matrix  $\Gamma$  is similar to the linear ma-

trix that was originally derived by Lagrange and Laplace to demonstrate that at first order, the Solar System is stable, while  $\Phi_3(w, \bar{w})$  and  $\Phi_5(w, \bar{w})$  gather the terms of degree 3 and 5.

The system of equations thus obtained contains some 150,000 terms, but can be considered as a simplified system, as its main frequencies are now the precession frequencies of the orbits of the planets, and no longer comprise their orbital periods. The full system can thus be numerically integrated with a very large stepsize of 200 to 500 yr. Contributions due to the secular perturbation of the Moon and general relativity are also included (see Laskar, 1990, 1996, for more details and references).

This secular system is then simplified and reduced to about 50,000 terms, after neglecting terms of very small value (Laskar, 1994). Finally, a small correction of the terms of the matrix  $\Gamma$  of (10), after diagonalization, is performed in order to adjust the linear frequencies, in a similar way as it was done in (Laskar, 1990). Indeed, in the outer planetary system, terms of higher order are of some importance, but their effect will mainly be slightly to modify the values of the main frequencies of the system. The correction that is done here is a simple way to correct for this effect.

The original solutions of Laskar (1990) are very close to the direct numerical integration La2003 over 10 to 20 Myr (Laskar et al., 2004). With the present small adjustment, we obtain a significant improvement, and the solutions are very close over about 35 Myr (Figs. 15, 16). It should be noted that this time is also about the time over which the numerical solution itself is valid (Fig. 4), due to the imperfections of the model. Moreover, as the stepsize used in the secular equations is 200 yr instead of 1.82625 days, over very long times the numerical noise will be smaller. It is thus legitimate to investigate the diffusion of the orbital motion over long times using the secular equations. The major advantage, besides reducing the roundoff errors, resides in the computation speed: the integration of the secular equations is 2000 times faster than the integration of the non-averaged equations, and we can compute a 5 Gyr solution for the Solar System in 12 hours on a Compaq alpha workstation (833 MHz). We will thus be able to make statistics over many solutions with close initial conditions. In these computations, our main limitation will be in the huge amount of data generated by these numerical integrations.

## 5.2. Eccentricity

We have integrated first 403 orbital solutions, retaining only the largest values reached by the eccentricity. The results of the maximum eccentricity reached after 25, 100, 250, 500, 1000, 2000, and 5000 Myr are given in Fig. 17. As expected in a diffusive process, the maximum value reached by the eccentricity increases slowly with time.

We have also performed 200 orbital solutions, retaining the values of the eccentricity with a sampling rate of 25,000 yr in order to establish the density distribution of

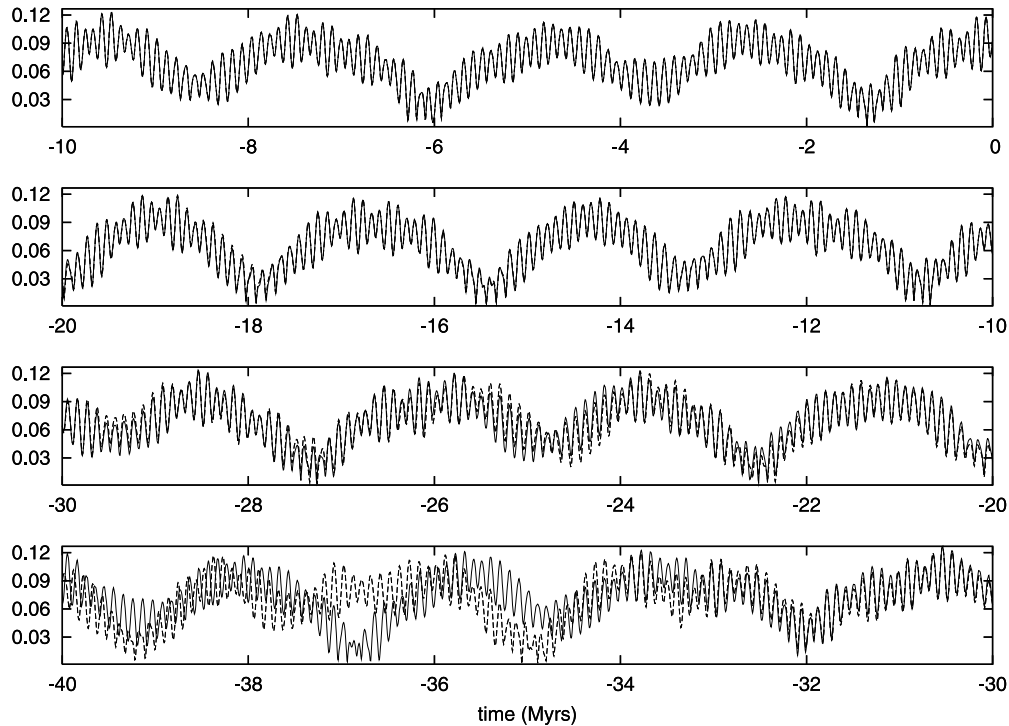


Fig. 15. Eccentricity of Mars over 40 Myr. The full line is the pure numerical solution La2003, while the dotted line is obtained with the secular equations from (Laskar, 1990, 1994) (Eq. (10)). The two solutions are quasi-identical over 32 to 35 Myr.

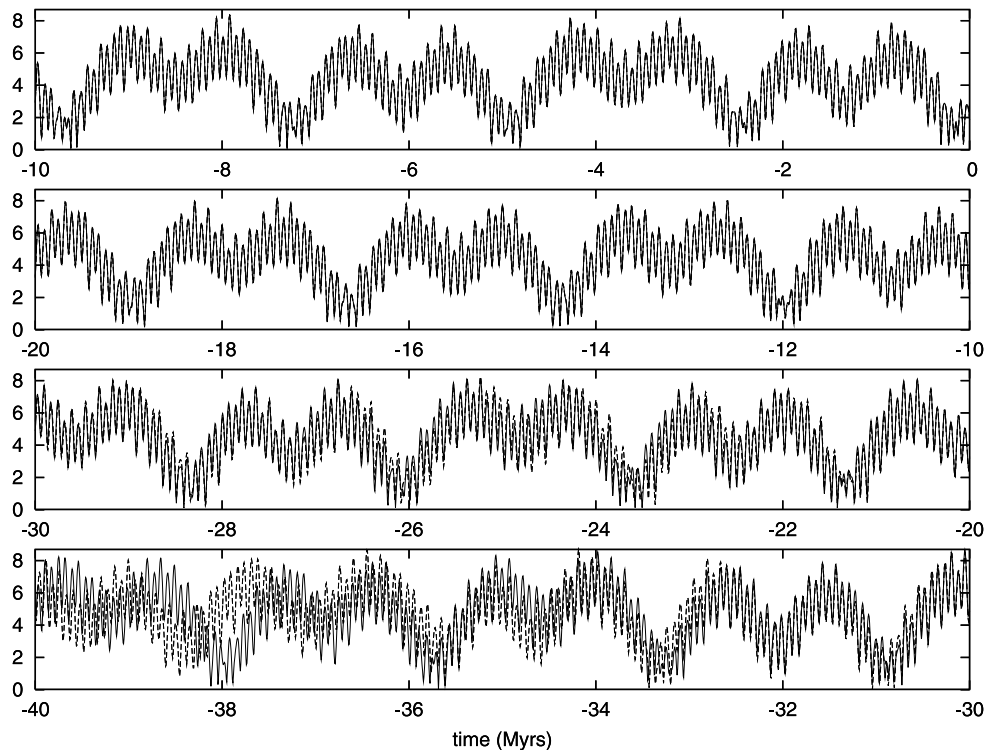


Fig. 16. Inclination of Mars over 40 Myr (in degrees, referred to the ecliptic J2000). The full line is the pure numerical solution La2003, while the dotted line is obtained with the secular equations from (Laskar, 1990, 1994) (Eq. (10)). The two solutions are quasi-identical over 32 to 36 Myr.

the eccentricity over different time spans (Fig. 18). It is striking that the density function converges towards a very smooth density function beyond 1 Gyr, as the chaotic diffu-

sion is dominant, while over 25 Myr, where the motion is very predictable, the density distribution cannot be approximated by a simple function. The associated mean values ( $\bar{e}$ )

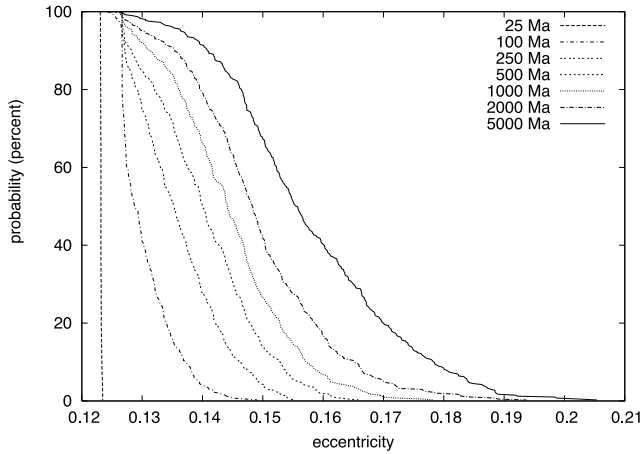


Fig. 17. The probability (vertical axis) of reaching a given eccentricity (horizontal axis) after a specified time is given by the different curves, for 25, 100, 250, 500, 1000, 2000, 5000 Myr. The estimate is obtained with 403 different orbital solutions of the secular equations.

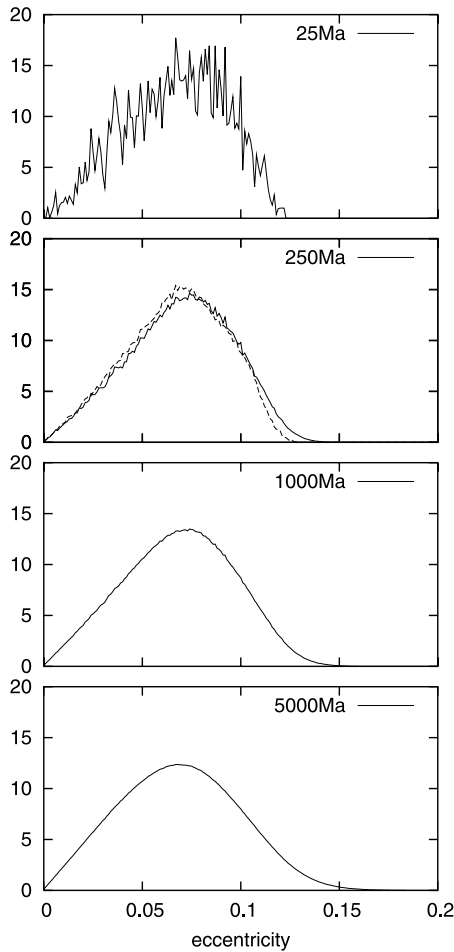


Fig. 18. Normalized density function for the eccentricity of Mars over 25, 250, 1000, and 5000 Myr. For 250 Myr, the dotted line is the density obtained over the pure numerical solution La2003.

and standard deviations ( $\sigma_e$ ) of the eccentricity distributions are given in Table 3.

Table 3  
Statistics for the eccentricity evolution over 5 Gyr

$T$ (Myr)	$\bar{e}$	$\sigma_e$
25	0.066765	0.025551
100	0.069046	0.025592
250	0.069664	0.026361
500	0.069430	0.026914
1000	0.068936	0.027725
2000	0.068754	0.028602
5000	0.068989	0.029937

400 solutions of the secular equations are integrated over 5 Gyr with close initial conditions.

Table 4  
Maximal and minimal values reached by the obliquity over 5 Gyr

$T$ (Myr)	$\varepsilon_M$ ( $^\circ$ )	$\varepsilon_m$ ( $^\circ$ )
5	38.442	13.594
10	46.907	13.594
25	50.310	13.594
50	62.274	3.786
100	65.874	0.014
250	70.508	0.001
500	72.543	0.001
1000	74.678	0.000
2000	76.005	0.000
3000	79.465	0.000
5000	82.035	0.000

200 solutions of the secular equations were integrated over 5 Gyr with close initial conditions, and for each orbital solution, 1001 obliquity solutions with initial conditions for regularly spaced values of  $p$  ranging from  $p_0 - \delta p/2$  to  $p_0 + \delta p/2$  were integrated.

In Fig. 18, even if the density function of the eccentricity over 5000 Gyr is very smooth, it appears to be slightly different from a true Gaussian function. In particular, the density goes to zero for zero eccentricity, with a linear behavior near the origin. This will be discussed in Section 5.4.1.

### 5.3. Obliquity

We have done the same analysis for the obliquity over 5 Gyr, with 1001 solutions of the obliquity, for each of the 200 orbital solutions, integrated with close initial conditions. The maximum obliquity reached after 10, 25, 50, 100, 250, 500, 1000, 2000, 3000, 5000 Myr are given in Tables 4 and 6 and in Fig. 19.

The density function for the obliquity over 50, 100, 250, 500, 1000, 2000, 3000, 4000 Myr is displayed in Fig. 20, and the corresponding mean values  $\bar{\varepsilon}$  and standard deviation  $\sigma_\varepsilon$  are given in Table 5.

The maximal value reached by the obliquity is  $82.035^\circ$ , although the probability to reach more than  $80^\circ$  over 5 Gyr is only 0.015% (Table 6). More realistically, there are still 4% of the solutions that went beyond  $70^\circ$  in less than 3 Gyr. In these cases, the orbits were able to find their way in the zone of chaotic small diffusion between the resonances  $s_1$  and  $s_7$  (Fig. 13). Finally, we are nearly certain that the obliquity of Mars went to higher values than  $60^\circ$  during its history,

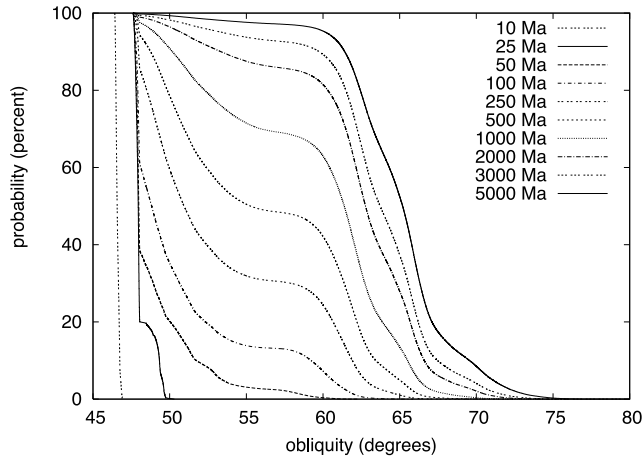


Fig. 19. The probability (vertical axis) of reaching a given obliquity (horizontal axis) after a specified time if given by the different curves, for 10, 25, 100, 250, 500, 1000, 2000, 3000, 5000 Myr. The estimate is obtained with 200 different orbital solutions, with, for each of them, 1001 integrations for regularly spaced values of  $p$  ranging from  $p_0 - \delta p/2$  to  $p_0 + \delta p/2$ .

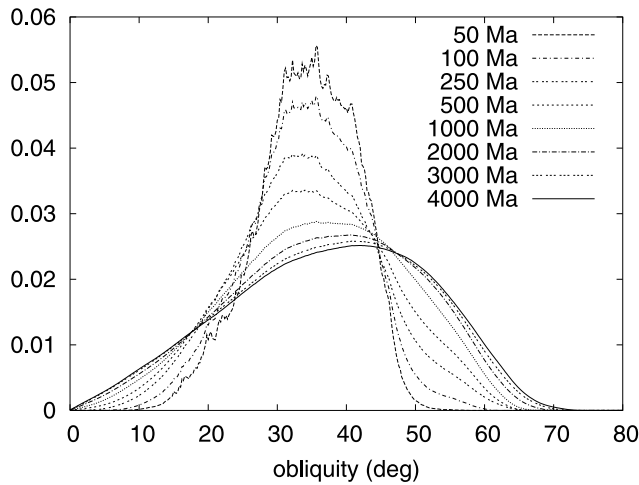


Fig. 20. Normalized density distribution of the obliquity of Mars, over 50, 100, 250, 500, 1000, 2000, 3000, and 4000 Myr. This is obtained with 19 different orbital solutions, and for each of them 501 obliquity solutions, with a sampling time of 1000 yr.

as 95% of the solutions reached this value. Closer to the present, the probability to have reached  $60^\circ$  before 1 Gyr is about 63%.

In Fig. 21, the similar quantities are plotted for a single solution of the obliquity. In this case, the density function of the obliquity is very different, as over 4 Gyr (full line), it has two distinct maxima. Indeed, the individual solutions can present very different behavior, and in the present case, it appears that the solution remained for a long time in the first “box”  $B1$  related to secular resonances with  $s_1$ , and also for a long period in the “box”  $B2$  related to  $s_2$ , but never much in between (see Fig. 13). When the statistics are made over many orbital solutions, these regions will be displaced, and the repartition of the obliquity will become more even.

Table 5

Mean value ( $\bar{\epsilon}$ ) and standard deviation ( $\sigma_\epsilon$ ) for the obliquity over 50 Myr to 4 Gyr

$T$ (Myr)	$\bar{\epsilon}$ ( $^\circ$ )	$\sigma_\epsilon$ ( $^\circ$ )
50	34.244	7.149
100	34.068	8.438
250	34.432	9.979
500	35.082	10.977
1000	35.950	11.987
2000	36.779	12.978
3000	37.267	13.499
4000	37.620	13.814

19 solutions of the secular equations were integrated over 4 Gyr with close initial conditions, and for each orbital solution, 501 obliquity solutions with initial conditions for regularly spaced values of  $p$  ranging from  $p_0 - \delta p/2$  to  $p_0 + \delta p/2$  were integrated.

Table 6

Numerical examples taken from the data of Fig. 19

$T$ (Myr)	$60^\circ$	$65^\circ$	$70^\circ$	$75^\circ$	$80^\circ$
5000	95.35	52.51	8.51	0.270	0.015
3000	89.29	37.38	4.07	0.025	
2000	81.47	27.94	2.08	0.005	
1000	63.00	13.52	0.36		
500	42.16	4.92	0.03		
250	24.40	1.03	0.01		
100	7.23				
50	0.36				
25					

For each time span  $T$ , we give respectively the probability (in percent) for the obliquity  $\epsilon$  to reach  $60^\circ$ ,  $65^\circ$ ,  $70^\circ$ ,  $75^\circ$ , or  $80^\circ$ . When there is no occurrence of the event, the corresponding space is left blank.

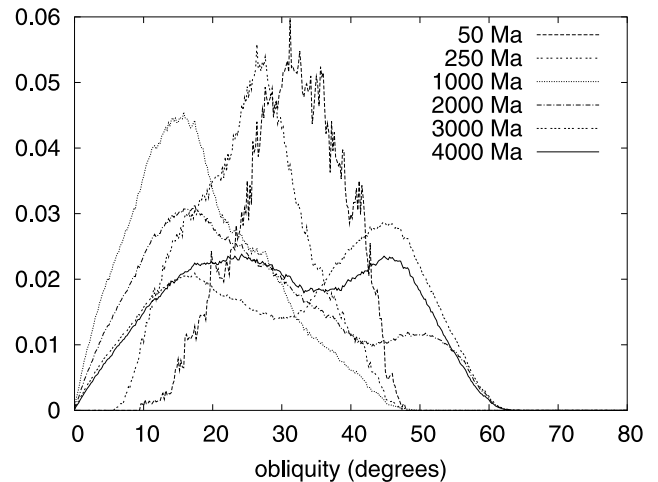


Fig. 21. Normalized density distribution of the obliquity of Mars, over 50, 250, 1000, 2000, 3000, and 4000 Myr. This is obtained with a single solution with a sampling time of 1000 yr.

#### 5.4. Diffusion laws

We have seen that the density functions of the eccentricity (Fig. 18) or obliquity (Fig. 20) over 5 Gyr are smooth functions, that evolve with time and differ significantly from pure Gaussian distributions. In this section, we will look to the



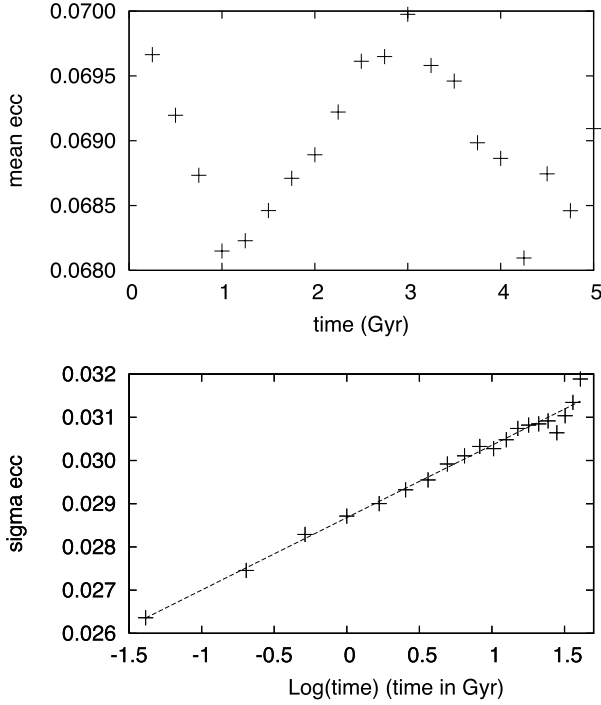


Fig. 22. Evolution of the mean value  $\bar{e}$  (top) and standard deviation  $\sigma_e$  (bottom) for the eccentricity of Mars. The statistics are made over 20 consecutive intervals of 250 Myr, for 200 orbits with close initial conditions. The standard deviation is fitted with the line  $\sigma_e = 0.028678 + 0.001676 \log(T)$  where the time  $T$  is in Gyr.

evolution with time of these distributions, and analyze the diffusion of the solutions over time. The goal is to establish some simple formulas that will allow to represent the distributions of the eccentricity and obliquity over very long times, of several Gyr. Quite remarkably, although it is impossible to predict the precise evolution of the individual trajectories, we will be able to give very simple expressions that fit very well with the observed eccentricity and obliquity distributions.

#### 5.4.1. Eccentricity

We first have looked at the evolution of the mean value of the eccentricity  $\bar{e}$  and its standard deviation  $\sigma_e$  over 5 Gyr, computing these quantities over 20 consecutive intervals of 250 Myr (Fig. 22). The mean value  $\bar{e}$  does not show significant evolution, but the standard deviation increases steadily and is well fitted with

$$\sigma_e = 0.028678 + 0.001676 \log(T), \quad (11)$$

where the time  $T$  is expressed in Gyr. In a pure Brownian motion, the diffusion speed  $d\sigma^2/dt$  is constant, while here, when neglecting the terms in  $\log^2(T)$  that are small over a few Gyrs, we have

$$\frac{d\sigma_e^2}{dT} \propto \frac{1}{T}. \quad (12)$$

The diffusion rate becomes thus very slow as  $T$  increases.

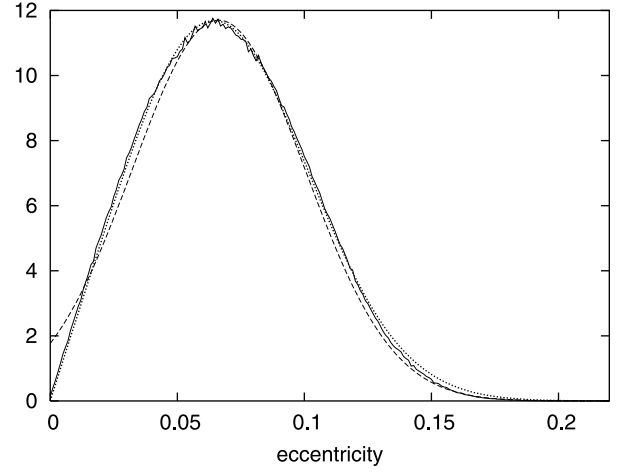


Fig. 23. Normalized density function for the eccentricity of Mars on the time interval [4.75 Gyr, 5 Gyr] (full line). The dashed line is the best fit Gaussian distribution, and the dotted line is the best fit with a distribution of a random walk with an absorbing end at 0 (Eq. (13)).

At the moment, we have no possibility to derive analytically the observed density function from the equation of motion, but we have searched for the best possible fit with a small number of parameters (Fig. 23). As was already observed, the Gaussian density is not a good candidate as the observed density goes to zero when the eccentricity goes to zero, with a non-zero derivative. In fact, a model that fits very well with the density of the eccentricity is given by a one-dimensional random walk with an absorbing edge at zero (see, for example, [Grimmett and Stirzaker, 2001](#)). For such a simple model, the density function is

$$\begin{aligned} \tilde{G}_{\tilde{m}, \tilde{\sigma}}(x) = & \left( \exp\left(-\frac{(x - \tilde{m})^2}{2\tilde{\sigma}^2}\right) - \exp\left(-\frac{(x + \tilde{m})^2}{2\tilde{\sigma}^2}\right) \right) \\ & \times \left( \sqrt{2\pi}\tilde{\sigma} \operatorname{erf}\left(\frac{\tilde{m}}{\sqrt{2}\tilde{\sigma}}\right) \right)^{-1}, \end{aligned} \quad (13)$$

where  $\operatorname{erf}(x)$  is the error function

$$\operatorname{erf}(x) = \frac{2}{\sqrt{\pi}} \int_0^x e^{-t^2} dt. \quad (14)$$

In Fig. 23, the density function  $f_e(x)$  of the eccentricity on the time interval [4.75 Gyr, 5 Gyr] is plotted in full line, while the dashed line is the best fit with a Gaussian density. Barely visible, as it superposes nearly exactly with the filled curve, is the best least square approximation of  $f_e(x)$  with  $\tilde{G}_{\tilde{m}, \tilde{\sigma}}(x)$ , where  $\tilde{m} = 0.063831$  and  $\tilde{\sigma} = 0.037436$ . It should be noted that the mean  $m_a$  and standard deviation  $\sigma_a$  for the density function  $\tilde{G}_{\tilde{m}, \tilde{\sigma}}(x)$  are not  $\tilde{m}$  and  $\tilde{\sigma}$ , but slightly different:

$$m_a = \frac{\tilde{m}}{\operatorname{erf}(\tilde{m}/\sqrt{2}\tilde{\sigma})} \quad (15)$$

and

$$\sigma_a^2 = \tilde{\sigma}^2 + \tilde{m}^2 - m_a^2 + \sqrt{\frac{2}{\pi}} \tilde{\sigma} m_a \exp\left(-\frac{\tilde{m}^2}{2\tilde{\sigma}^2}\right). \quad (16)$$

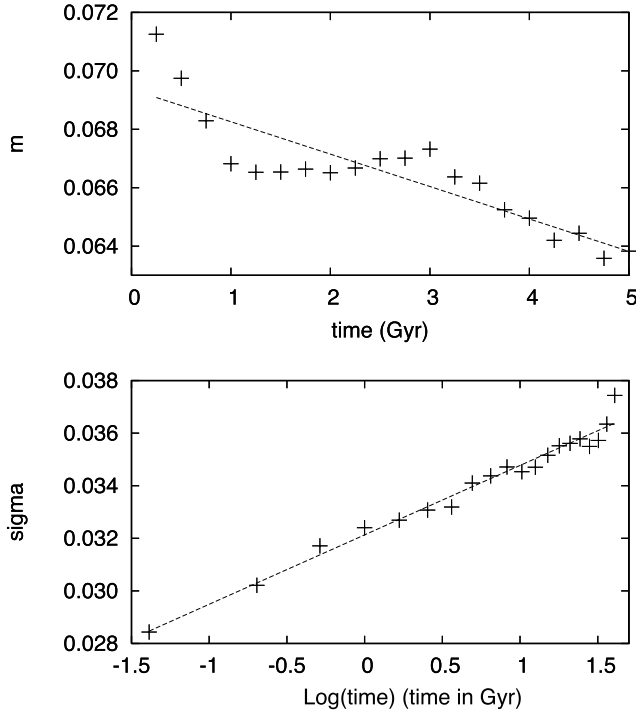


Fig. 24. Evolution with time of the parameters  $\tilde{m}$  (top) and  $\tilde{\sigma}$  (bottom) of the eccentricity density distribution given in (Eq. (13)). The linear fits are made respectively with  $\tilde{m} = 0.069367 - 0.001109T$  and  $\tilde{\sigma} = 0.032131 + 0.002641 \log(T)$ , where the time  $T$  is in Gyr.

As this density  $\tilde{G}_{\tilde{m},\tilde{\sigma}}(x)$  represents very precisely the evolution of Mars' eccentricity from 500 Myr to 5 Gyr, we have computed the corresponding parameters  $\tilde{m}, \tilde{\sigma}$  (Fig. 24) that can be approximated over [0.5 Gyr, 5 Gyr] by

$$\begin{aligned} \tilde{m} &= 0.069367 - 0.001109T, \\ \tilde{\sigma} &= 0.032131 + 0.002641 \log(T), \end{aligned} \quad (17)$$

where the time  $T$  is in Gyr. In some sense, the relations (13) and (17) provide a good predictive model for the eccentricity of Mars beyond 500 Myr.

#### 5.4.2. Obliquity

We have performed the same study for Mars' obliquity. In this case, the analysis was performed for 19 orbital solutions, and for each of them 500 obliquity solutions with close initial conditions. The statistics were made over 40 consecutive intervals of 100 Myr. Contrarily to the eccentricity, the mean obliquity  $\bar{\varepsilon}$  presents a slow increase with time as well as its standard deviation that can be well approximated (Fig. 25) with

$$\begin{aligned} \bar{\varepsilon} &= 37.007 + 1.345 \log(T), \\ \sigma_{\varepsilon} &= 13.301 + 1.426 \log(T) - 0.259 \log^2(T). \end{aligned} \quad (18)$$

The diffusion of the obliquity thus follows a similar law as (12) for the eccentricity. In a similar way as with the eccentricity, we have plotted the last density function  $f_{40}(\varepsilon)$ , obtained over the interval [3.9 Gyr, 4.0 Gyr] (Fig. 27, full line). The density  $f_{40}(\varepsilon)$  is also far from a

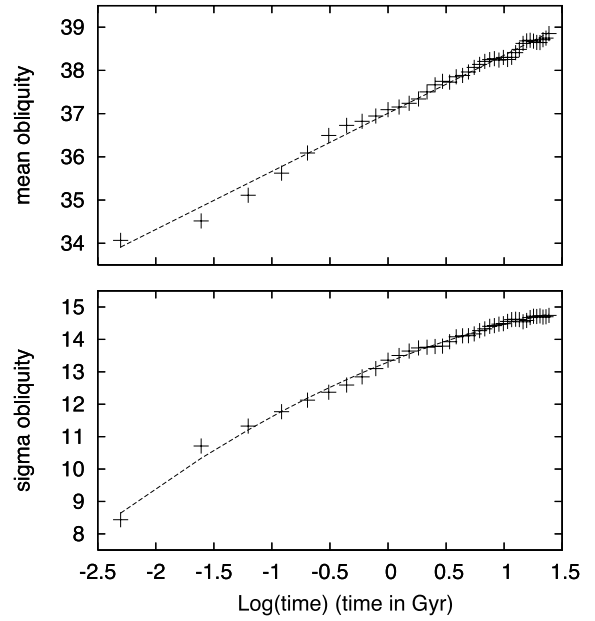


Fig. 25. Evolution of the mean value  $\bar{\varepsilon}$  (top) and standard deviation  $\sigma_{\varepsilon}$  (bottom) for the obliquity of Mars. The statistics are made over 40 consecutive intervals of 100 Myr, for about 10,000 orbits with close initial conditions. The mean obliquity is fitted with  $\bar{\varepsilon} = 37.007 + 1.345 \log(T)$ , while the standard deviation is fitted with  $\sigma_{\varepsilon} = 13.301 + 1.426 \log(T) - 0.259 \log^2(T)$ , where  $T$  is in Gyr.

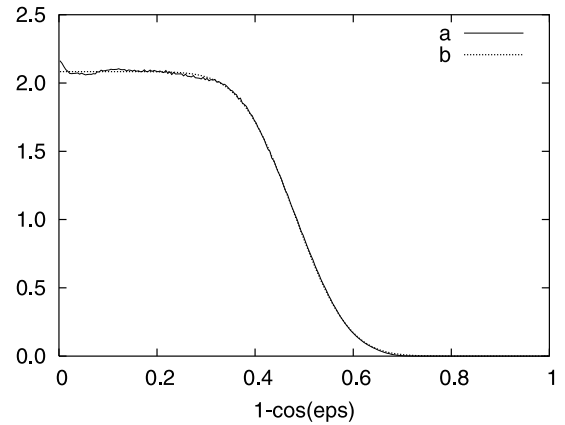


Fig. 26. Normalized density function for  $1 - \cos(\varepsilon)$  (full line, (a)), where  $\varepsilon$  is the obliquity of Mars on the time interval [3.9 Gyr, 4 Gyr]. The dotted line (b) the best fit with the density function given in Eq. (19).

Gaussian density (dashed line), although the density  $f_1(\varepsilon)$  for [0 Myr, 100 Myr] is very similar to a Gaussian density. In fact the obliquity density of (Fig. 27) is also different from the density of the eccentricity (Eq. (13)). After some trials, we found a very good fit of  $f_{40}(\varepsilon)$  with a sine function on  $[0^\circ, 40^\circ]$ . This led us to search for the density function  $f_Y$  of  $Y = 1 - \cos(\varepsilon)$  instead of  $\varepsilon$  (Fig. 26). We have found a very good agreement of the observed density  $f_Y$  with

$$f_{m,b}(x) = \frac{1}{2m} \{ \operatorname{erf}(b(x+m)) - \operatorname{erf}(b(x-m)) \}. \quad (19)$$

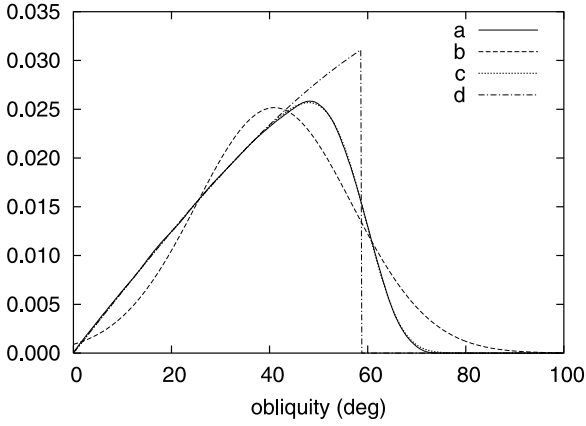


Fig. 27. Normalized density function for the obliquity ( $\varepsilon$ ) of Mars on the time interval [3.9 Gyr, 4 Gyr] (full line, (a)). The dashed line (b) is the best fit Gaussian distribution, and the dotted line (c) the best fit with the density function given in (Eq. (23)). The curve (d) is the density (26) of the uniform probability on the spherical cap limited by  $\varepsilon = \varepsilon_{\hat{m}}$ .

This agreement is in fact so precise that the two curves are barely distinguishable on Fig. 26. The density  $f_{m,b}(x)$  is in fact the density of  $|X + Y|$ , where  $X, Y$  are two independent random variables. The random variable  $X$  has a uniform density on  $[-m, m]$ , and  $Y$  is a centered Gaussian variable with standard deviation

$$\sigma = \frac{1}{\sqrt{2b}}. \quad (20)$$

In fact,  $f_{m,b}(x)$  is a density on  $[0, +\infty[$ , and not on  $[0, 2]$ . Nevertheless, for the values of  $b, \hat{m}$ , that we will use ( $b \approx 8, \hat{m} \approx 0.5$ ), and for  $x > 2$ ,

$$f_{m,b}(x) \simeq \frac{2b}{\sqrt{\pi}} \exp(-b^2 x^2) \quad (21)$$

and

$$\int_0^2 f_{\hat{m},b}(x) dx \simeq \text{erf}(2b). \quad (22)$$

For  $b = 8$ ,  $1 - \text{erf}(2b) \simeq 0.23 \times 10^{-112}$ . We can thus consider, for our practical needs, that  $f_{m,b}(x)$  is a density on  $[0, 2]$ . Once we have a good candidate for the density of  $1 - \cos(\varepsilon)$ , we obtain easily the density of  $\varepsilon$  on  $[0, \pi]$  as

$$\hat{G}_{\hat{m},b}(x) = \frac{\sin(x)}{2\hat{m}} \left\{ \text{erf}(b[1 - \cos(x) + \hat{m}]) - \text{erf}(b[1 - \cos(x) - \hat{m}]) \right\}. \quad (23)$$

The parameters  $\hat{m}, b$  are determined by least square fit to the observed density of the obliquity, and we find  $\hat{m}_{40} \simeq 0.479, b_{40} \simeq 7.793$ . As with the eccentricity, the agreement is so good that  $\hat{G}_{\hat{m},b}(x)$ , plotted with a dotted line (label c) in Fig. 27 is barely discernible from the density of the obliquity.

We have then determined the parameters  $\hat{m}, b$  on each time interval, but instead of plotting  $\hat{m}$  in Fig. 28, we have preferred to display the corresponding value of the obliquity

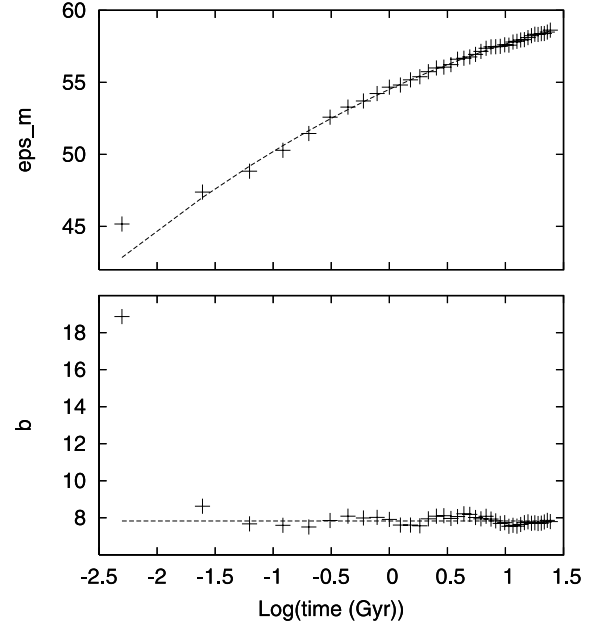


Fig. 28. Evolution with time of the parameters  $\hat{m}$  (top) and  $b$  (bottom) of the obliquity density distribution given in Eq. (23). The least square fit of  $\hat{m}$  is  $\hat{m} = 54.515 + 3.726 \log(T) - 0.583 \log^2(T)$ . The mean value of  $b$ ,  $b_m = 7.825$  is plotted in dotted line.

( $\varepsilon_{\hat{m}} \simeq 58.622^\circ$  in Fig. 27), given by

$$\varepsilon_{\hat{m}} = \arccos(1 - \hat{m}). \quad (24)$$

The value of  $b$  is nearly constant, with mean value  $\bar{b} = 7.825$ , while  $\varepsilon_{\hat{m}}$  can be approximated by

$$\varepsilon_{\hat{m}} = 54.515 + 3.726 \log(T) - 0.583 \log^2(T), \quad (25)$$

where  $T$  is in Gyr. As for the eccentricity, Eqs. (23) and (25) provide a precise description of the obliquity evolution beyond 500 Myr. It should be noted that the density (Fig. 27d)

$$f_{\hat{m}}(x) = \frac{\sin(x)}{\hat{m}} 1_{[0, \varepsilon_{\hat{m}}]}(x), \quad (26)$$

where  $1_{[0, \varepsilon_{\hat{m}}]}(x)$  is the characteristic function of the interval  $[0, \varepsilon_{\hat{m}}]$ , represents also the uniform density on a spherical cap centered on the normal to the orbital plane of Mars, and limited by  $\varepsilon = \varepsilon_{\hat{m}}$ , with the usual Lebesgue measure on the sphere.

We can now look back to the evolution of the density with time given in Figs. 14 and 20. Over a short time (50 Myr), the obliquity is localized around a mean value of about  $34^\circ$ , with a Gaussian density; but as the diffusion process evolves, the axis density function will evolve towards a uniform distribution on the spherical cap limited by  $\varepsilon = \varepsilon_{\hat{m}}$ , where  $\varepsilon_{\hat{m}}$  is given by Eq. (25) or Fig. 28, with an additional random noise. The distribution of the action variable  $X = L \cos \varepsilon$ , or equivalently of the precession frequency ( $\propto \cos \varepsilon$ ), will also be uniform with the addition of a random noise.

## 6. Conclusions

The new orbital and obliquity solution presented here can be used over 10 to 20 Myr for precise paleoclimate studies on Mars. It is important to recall that the severe increase of obliquity at 5 Myr is robust with respect to the uncertainty on the model and initial conditions. It thus provides a constraint on the possible past evolution of the martian climates, and in particular on the evolution of the ice caps (Laskar et al., 2002). The full solution, together with subroutines allowing the computation of the insolation for various latitudes, derived from the similar routines used for the Earth paleoclimate studies (Laskar et al., 1993) is available on the WEB at <http://www.imcce.fr/Equipes/ASD/mars.html>.

The orbital solution of Mars should be precise over about 40 Myr, much longer than the time of validity of the obliquity and precession quantities. We have provided as well the nominal orbital solution of Mars over 100 Myr, for reference, and as an example of possible evolution, although we know that with the problem of numerical roundoff error alone, the solution has practically no chance to be valid over more than 60 Myr.

Over longer time scale, beyond 100 Myr, the chaotic regime prevails, and we cannot give any precise evolution of the obliquity or of the orbit. Nevertheless, we are able to give here a very precise estimate of the density function of the obliquity and eccentricity over the age of the Solar System, and have thus in a very concise way all the information for producing statistical estimates of the past climate evolution of Mars along its history. In particular, beyond 500 Myr, the density distribution of the axis of Mars tends towards a uniform density on a spherical cap limited by  $\varepsilon = \varepsilon_{\hat{m}}$  (Eq. (25)), with the addition of a random noise that produces a slow diffusion of the obliquity beyond  $\varepsilon_{\hat{m}}$ .

The computations beyond 250 Myr are made with the averaged equations of Laskar (1990), but this should not modify significantly the results from a similar study made with non-averaged equations as the two solutions are very similar over 35 Myr (Figs. 15 and 16). The chaotic behavior of the inner planets mostly results from their secular interactions (Laskar, 1990), and the statistics made over the overlapping 250 Myr on the obliquity are very similar (Tables 2 and 5).

It is remarkable that the present value of the obliquity of Mars ( $\varepsilon_0 \approx 25.19^\circ$ ) is very far from its mean value, evaluated over 4 Gyr ( $\bar{\varepsilon} \approx 37.62^\circ$ ); it is even further from its most probable value over 4 Gyr,  $\varepsilon_s \approx 41.80^\circ$  obtained for the maximum of the 4 Gyr density of Fig. 20. The situation is different for the eccentricity, as the mean ( $\bar{e} \approx 0.069$ ) and most probable value ( $e_s \approx 0.068$ ) are smaller than the present value ( $e_0 \approx 0.093$ ). In the search of the past climatic evolution of Mars, one could thus define the “standard model” of Mars as the most probable one, with  $e_s = 0.068$ ,  $\varepsilon_s = 41.80^\circ$ .

This suggests that the present dry and cold martian climate is probably not representative of the past current environmental conditions. Climatic simulations with orbital

parameters close to the “standard model” shows that large increases in summer insolation may cause dynamical instability of the polar caps and an intensive sublimation multiplying global atmospheric humidity by a factor  $\sim 50$  (Jakosky et al., 1995; Richardson and Wilson, 2002; Mischna et al., 2003). If not limited by a thermally protecting dust lag, dramatic annual water loss (estimated close to  $10 \text{ cm yr}^{-1}$ ) would lead to a quick disappearance of polar caps over some obliquity cycles and to deposition of stable surface ice in equatorial areas (Jakosky et al., 1995). Geomorphological features should illustrate this now “current” latter situation. We can expect that conjugated efforts in climate modeling (e.g., Haberle et al., 2000; Mischna et al., 2003), geomorphological observations from present and future spacecraft missions (e.g., Head and Marchant, 2003) and comparison with astronomical solutions will improve the reconstruction of the past “high obliquity” martian climate.

## Acknowledgments

J.L. thanks A. Coquio and B. Derrida for discussions on random processes. A large part of the computations were made at IDRIS-CNRS, and we benefited from support from PNP-CNRS, and CS from Paris Observatory.

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